Aircraft	Information
Aircraft Registration	G-IBIG
Aircraft Type	Bell 206B
Aircraft Serial Number	2202
Airworthiness Review	8244.1
Aircraft Datcon Reading	N/A
Engine #1 Type	RR M250-C20
Engine #1 Serial Number	CAE-830068
Engine #1 TTSN	4135.9
Engine #1 Cycles	632
Engine #1 TSO	N/A
Torque events	N/A
Landings	N/A
Year of construction	1977
Current Date	19/02/2024
	MP/05017/P
Maintenance Programme	Issue 01 Rev 00
	23/01/2024

Aircraft Status Report

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Last	Complied V	Vith		Next Due	-		Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
				Sch	neduled Airf	rame Inspect	ions							
Thoroughly clean swa solvent (C-304) to rem wipe dry. Visually insp cracks with a 10X map side of swashplate sup assembly. If a crack is inspection.	shplate support base a love all dirt, grease, an bect full circumference gnifying glass. Pay part oport suspected carry out fu	rea with drycleaning d oil residue, then of fillet radius for icular attention to aft rther dye penetrant	50	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A
Remove cabin overhead blanket, aft vertical tu and roll-over bulkhead	move cabin overhead upholstery, hat bin, soundproofing anket, aft vertical tunnel covers, d roll-over bulkhead access panels.			N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
Remove roof beam ac	nove roof beam access panels.			N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
Examine flight control security.	nove roof beam access panels. mine flight control tubes inside roof beam for condition ar urity.			N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
Examine vertical tunn condition.	el, roof beam, and roo	f shell structure for	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
Examine aft passenge and indication of scre nutplates.	r seat bulkhead for dis w contact in the area c	tortion and/or cracks f hat bin attachment	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
Examine electrical co security, and cleanline	mponents behind hat ess.	bin for condition,	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
Examine engine and to cracks, corrosion, or lo	ransmission support st pose fasteners.	ructure for I	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Las	t Complied V	Vith		Next Due	-		Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
In the vertical tunnel a tubes for evidence of	area, examine the fligh chafing with oil line or	nt control system flex cable.	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
In the vertical tunnel a bellcranks, levers, sup excessive looseness, a	area, examine the fligh ports, and walking be nd security of attachn	nt control system ams for binding, nent.	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
In the vertical tunnel a condition, and securit	the vertical tunnel area, examine the oil lines for leakage, ndition, and security. The vertical tunnel area, examine the roof beam interface tical tunnel for cracks, corrosion, and loose fasteners.				N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
In the vertical tunnel a vertical tunnel a	the vertical tunnel area, examine the roof beam interface v tical tunnel for cracks, corrosion, and loose fasteners.				N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
In the vertical tunnel angles at roof beam ir fasteners.	ical tunnel for cracks, corrosion, and loose fasteners. he vertical tunnel area, examine the vertical tunnel stiffer les at roof beam interface for cracks, condition, and loose eners.				N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
Remove pilot and cop	ilot seat cushions and	panels.	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
Under the pilots and o throttle cable, bellcra excessive looseness, a	co-pilots seats inspect nks, levers, yokes, and nd security of attachn	the control tubes, N1 supports for bending, nent.	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
Under the pilots and o chafing with controls,	r the pilots and co-pilots seats inspect the control tub tle cable, bellcranks, levers, yokes, and supports for b sive looseness, and security of attachment. r the pilots and co-pilots seats inspect the heater duct ig with controls, if installed			N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
Under the pilots and o leakage and general c	co-pilots seats inspect ondition.	the oil lines for	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Las	t Complied V	Vith		Next Due			Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Under the pilots and o electrical components	co-pilots seats inspect to for condition and sec	the wiring and urity.	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
Under the pilots and o corrosion, cracks, and	co-pilots seats inspect to condition.	the structure for	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
Refit pilot and copilot	fit pilot and copilot seat cushions and panels. rify minimum friction of collective control and readjust as quired. Verify minimum friction of copilot collective contro				N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
Verify minimum friction required. Verify minin stick, if installed.	rify minimum friction of collective control and readjust as quired. Verify minimum friction of copilot collective control ck, if installed.				N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
Verify minimum frictio Verify minimum frictio	uired. Verify minimum friction of copilot collective control k, if installed. ify minimum friction of cyclic control and readjust as requi ify minimum friction of copilot cyclic control stick, if instal				N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
Verify minimum friction as required. Verify m control pedals, if insta	on of anti-torque conti inimum friction of cc illed.	rol pedals and readjust opilot anti-torque	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
Inspect latches on cre	w doors for correct adj	justment.	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
Inspect latches on pas adjustment.	senger doors, and bag	gage door for correct	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
Remove main rotor bl installed.	ade leading edge erosi	on protection tape, if	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Las	t Complied V	Vith		Next Due	-		Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Examine leading edge contour as necessary.	of blades for corrosion	n and erosion Restore	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
If applicable, reinstall protection tape.	main rotor blade lead	ng edge erosion	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
Verify and adjust swa	rify and adjust swashplate tilt friction as required. ue must be between 15 to 32 pounds (67 to 142 N). If out o its. adjust to 15 to 32 pounds (67 to 142 N) and check agai				N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
Value must be betwee limits, adjust to 15 to	en 15 to 32 pounds (67 32 pounds (67 to 142 l	to 142 N). If out of N) and check again.	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
Remove hydraulic pur	ts, adjust to 15 to 32 pounds (67 to 142 N) and check again				N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
Examine hydraulic pu and rotor tachometer security.	mp splines, transmissio and adapter splines fo	on oil pump splines, or condition and	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
Refit hydraulic pump	and rotor tachometer	generator.	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
Perform operational c	nydraulic pump and rotor tachometer generator. m operational check of turbine outlet temperature sys			N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
Replace airframe fuel 01103-1, or element 7	filter element, kit KD6 582301, as applicable.	51511, element 52-	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Las	t Complied V	Vith		Next Due			Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Remove outlet port fi 0.020 inch (0.508mm port side of filter head (applicable to fuel fi 621-001, -003, and -	tting from airframe fue) wire, pass wire throu d to ensure that hole is lter assemblies 30600 101, 52-2889-016, an	el filter head. Using a gh bleed hole in outlet : unobstructed 11, 306005, 222-366- d 52-2889-016A).	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
Purge airframe fuel fil	ter and engine fuel sys	stem.	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
Purge the engine fuel system purging proce	system per the Rolls-R dure.	oyce engine fuel	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
During ground run, op 2 minutes to ensure ru and to check for fuel l	perate helicopter at 10 emaining air is purged eaks.	0% № for minimum of from fuel filter head	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
Check engine driven f follows: A: Disconnec differential pressure s fuel boost pumps and illuminates. C: If the e turn OFF both fuel bo the fuel hose removed caution light does not impending bypass ca	nring ground run, operate helicopter at 100% № for minimum ninutes to ensure remaining air is purged from fuel filter head d to check for fuel leaks. eck engine driven fuel pump filter impending bypass functior lows: A: Disconnect the fuel hose from forward end of ferential pressure switch and plug the hose. B: Turn ON both el boost pumps and verify if engine FUEL FILTER caution light uminates. C: If the engine FUEL FILTER caution light illuminate rn OFF both fuel boost pumps. Remove the plug and reconne e fuel hose removed in step A. D. If the engine FUEL FILTER ution light does not illuminate, troubleshoot and correct pending bypass caution system.				N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
Remove starter gener condition.	ator. Examine mountir	ng pad and clamp for	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
Examine starter gener generator if not smoo excessive wear.	rator commutator for v th and bright, or if sho	vear. Replace starter wing	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Las	Complied V	Vith		Next Due	-		Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Inspect starter genera	tor brushes for wear.		300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
Examine starter gener condition and security	ator cooling duct and /.	retention clamp for	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
Examine the bearings rotation. Examine the and the mating spline spline starter generate	xamine the bearings of the starter generator for smooth otation. Examine the splines on the starter generator driveshaft nd the mating splines on the engine gear shaft for wear. If dry pline starter generator installed, clean splines and lubricate. Open access door in overhead baggage compartment. Open ngine compartment cowling. Visually inspect longeron and				N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
Open access door in o engine compartment engine pan interface f inspect exposed uppe between aluminum lo sealant coating should acceptable. d. If there surface seal is present compartment cowling surface seal is not pre procedures.	pline starter generator installed, clean splines and lubricate. Open access door in overhead baggage compartment. Open ingine compartment cowling. Visually inspect longeron and ingine pan interface for corrosion using bright light and mirror, inspect exposed upper and lower edges of the joints and fastene ietween aluminum longeron and titanium engine pan. Edge of ealant coating should be visible. No indications of corrosion are icceptable. d. If there is no indication of corrosion and the bond urface seal is present, replace access panel and close engine ompartment cowling door. If corrosion is present or bonded urface seal is not present, refer to the BHT-206-SRM-1 for repai				N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
Inspect engine mount condition.	attachment structures	s for cracks and	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
Remove fuel shutoff v corrosion, cracks, and security and conditior	ect engine mount attachment structures for cracks and ition. ove fuel shutoff valve access panel. Inspect area for sion, cracks, and fuel leaks. Check wiring and plumbing f ity and condition. Reinstall access panel after completion			N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
Inspect aft passenger contact or cracks in ar	rear bulkhead for indic ea of hat bin attachme	cation of screw ent anchor nuts.	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Las	t Complied V	Vith		Next Due	-		Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Inspect aft passenger bin attachment ancho	rear bulkhead for disto r nuts.	ortion in area of hat	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
Using dye penetrant n 206-061-432-031 for o mounting flange when be accomplished with	nethods, inspect oil co cracks. Give particular a re individual blades att impeller installed.	oler blower impeller attention to the ach. Inspection may	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
Torque check tail roto pounds (5.6 to 7.9 Nm	r gearbox retaining nu ı).	ts 50 to 70 inch-	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
During ground run, ch blade assembly and a	eck dynamic balance c djust as required.	of tail rotor hub and	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
600 Hours of compon	de assembly and adjust as required. Hours of component operation main transmission inspec				N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A
Inspect yoke and trun	nion bearing surfaces	for brinelling.	1200	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A
Inspect yoke, latch bo blocks, grips, and pitc	lts, strap pins, strap fit h horns for corrosion.	tings, trunnion, pillow	1200	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A
Visually inspect all hu damage.	b components for exce	essive wear or	1200	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A
Carry out inspection o	f main transmission.		1200	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Las	t Complied V	Vith		Next Due	-		Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Carry out inspection o	f long tail rotor drives	haft assembly.	1200	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A
Cyclic control stick. Re control sticks.	emove pilot and copilo	t, if installed, cyclic	1200	N/A	N/A	7967.6	20/04/07	N/A	9167.6	N/A	N/A	923.5	N/A	N/A
Cyclic control stick. Cl cyclic stick tube(s) and	clic control stick. Clean lower 4 inches (101.6 mm) of remov clic stick tube(s) and inspect. clic control stick. Visually inspect cyclic stick tubes for crack ch a 3X magnifying glass. Pay particular attention to the are facent to the two slots in the tube where two bolts secure t				N/A	7967.6	20/04/07	N/A	9167.6	N/A	N/A	923.5	N/A	N/A
Cyclic control stick. Vi with a 3X magnifying adjacent to the two sl tube when installed ir suspected, remove pa cyclic stick tube(s) for method. If a crack is for serviceable tube.	clic stick tube(s) and inspect. relic control stick. Visually inspect cyclic stick tubes for cracks ith a 3X magnifying glass. Pay particular attention to the area ljacent to the two slots in the tube where two bolts secure th be when installed in the pivot lever assembly. If a crack is spected, remove paint and thoroughly inspect lower end of t clic stick tube(s) for cracking using the fluorescent penetrant ethod. If a crack is found, replace the cyclic stick tube with a rviceable tube. rclic control stick. If no crack is found, inspect cyclic stick tube				N/A	7967.6	20/04/07	N/A	9167.6	N/A	N/A	923.5	N/A	N/A
Cyclic control stick. If mechanical and corro must be separated by damage is not to exce tube circumference. C inch (0.0635 mm) dep repair, and 1/3 tube c tube that does not me	spected, remove paint and thoroughly inspect lower end of t clic stick tube(s) for cracking using the fluorescent penetrant ethod. If a crack is found, replace the cyclic stick tube with a rviceable tube. clic control stick. If no crack is found, inspect cyclic stick tube echanical and corrosion damage. Damaged and repaired area ust be separated by 1.0 inch (25 mm) minimum. Mechanical mage is not to exceed 0.005 inch (0.127 mm) depth and 1/3 be circumference. Corrosion damage is not to exceed 0.0025 ch (0.0635 mm) depth before, and 0.005 inch (0.127 mm) afte pair, and 1/3 tube circumference. Condemn as unserviceable be that does not meet these criteria.				N/A	7967.6	20/04/07	N/A	9167.6	N/A	N/A	923.5	N/A	N/A
Cyclic control stick. In where stick tube is ins damage. Bore damage 1/4 of the circumferen replace lever assembl	chanical and corrosion damage. Damaged and repaired ar st be separated by 1.0 inch (25 mm) minimum. Mechanica nage is not to exceed 0.005 inch (0.127 mm) depth and 1/ e circumference. Corrosion damage is not to exceed 0.002 i (0.0635 mm) depth before, and 0.005 inch (0.127 mm) a air, and 1/3 tube circumference. Condemn as unserviceab e that does not meet these criteria. ic control stick. Inspect bore of cyclic pivot lever assembly re stick tube is installed for mechanical and corrosion nage. Bore damage is not to exceed 0.002 inch (0.051 mm) of the circumference. Limit of one repair per bore. Repair ace lever assembly as required.				N/A	7967.6	20/04/07	N/A	9167.6	N/A	N/A	923.5	N/A	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description	-		Periodity		Las	t Complied V	Vith		Next Due			Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Cyclic control stick. Fo polish out any accepta 600 grit abrasive cloth	or cyclic stick tubes con able damage using 400 o or paper (C-423).	isidered serviceable, to	1200	N/A	N/A	7967.6	20/04/07	N/A	9167.6	N/A	N/A	923.5	N/A	N/A
Cyclic control stick. Th and mild detergent to and developer. Dry pa	oroughly clean cyclic s completely remove rt completely.	tick tube with water residual penetrant	1200	N/A	N/A	7967.6	20/04/07	N/A	9167.6	N/A	N/A	923.5	N/A	N/A
Cyclic control stick. Ap metal area.	rclic control stick. Apply chemical film material (C-100) to be etal area. rclic control stick. 11. Touch up area with epoxy polyamide imer (C-204). Where finish paint coat is required, match			N/A	N/A	7967.6	20/04/07	N/A	9167.6	N/A	N/A	923.5	N/A	N/A
Cyclic control stick. 11 primer (C-204). Where to original finish with	clic control stick. 11. Touch up area with epoxy polyamide imer (C-204). Where finish paint coat is required, match original finish with polyurethane coating (C-245).				N/A	7967.6	20/04/07	N/A	9167.6	N/A	N/A	923.5	N/A	N/A
Cyclic control stick. In control sticks.	ner (C-204). Where finish paint coat is required, match original finish with polyurethane coating (C-245). lic control stick. Install pilot and copilot, if installed, cyclic strol sticks.			N/A	N/A	7967.6	20/04/07	N/A	9167.6	N/A	N/A	923.5	N/A	N/A
Remove, disassemble	, and clean main rotor	mast.	1500	N/A	N/A	7967.6	20/04/07	N/A	9467.6	N/A	N/A	1223.5	N/A	N/A
Visually inspect main and wear. Indication o check.	rotor mast splines for l of wear requires an ove	burrs, nicks, cracks, er pins dimensional	1500	N/A	N/A	7967.6	20/04/07	N/A	9467.6	N/A	N/A	1223.5	N/A	N/A
Visually inspect main corrosion. Inspect sur	Illy inspect main rotor mast spines for burrs, nicks, crai vear. Indication of wear requires an over pins dimensio «. Illy inspect main rotor mast inner and outer surfaces fo sion. Inspect surface protective coatings for condition.		1500	N/A	N/A	7967.6	20/04/07	N/A	9467.6	N/A	N/A	1223.5	N/A	N/A
Visually inspect main erosion, spalling, and	rotor mast bearing bal brinelling.	ls and races for pits,	1500	N/A	N/A	7967.6	20/04/07	N/A	9467.6	N/A	N/A	1223.5	N/A	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Las	t Complied V	Vith		Next Due	-		Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Reassemble and insta	ll main rotor mast.		1500	N/A	N/A	7967.6	20/04/07	N/A	9467.6	N/A	N/A	1223.5	N/A	N/A
Remove main transmi	ssion from aircraft.		1500	N/A	N/A	7967.6	20/04/07	N/A	9467.6	N/A	N/A	1223.5	N/A	N/A
Remove mast assemb main transmission. Ins contamination.	emove mast assembly, oil filter, and housing assembly from ain transmission. Inspect oil filter for debris and metal ntamination. emove top case, planetary, and sun gear from main ansmission.				N/A	7967.6	20/04/07	N/A	9467.6	N/A	N/A	1223.5	N/A	N/A
Remove top case, plan transmission.	netary, and sun gear fr	om main	1500	N/A	N/A	7967.6	20/04/07	N/A	9467.6	N/A	N/A	1223.5	N/A	N/A
Inspect main transmis and accessible areas o attention to the spline	nsmission. pect main transmission top case, planetary assembly, sun ge d accessible areas of the gearshaft for condition. Pay particu ention to the splines of the sun gear and gearshaft.				N/A	7967.6	20/04/07	N/A	9467.6	N/A	N/A	1223.5	N/A	N/A
Install main transmiss	ion sun gear, planetar	y, and top case.	1500	N/A	N/A	7967.6	20/04/07	N/A	9467.6	N/A	N/A	1223.5	N/A	N/A
Install mast, oil filter o	onto main transmissio	ı.	1500	N/A	N/A	7967.6	20/04/07	N/A	9467.6	N/A	N/A	1223.5	N/A	N/A
Refit main transmissic	Il mast, oil filter onto main transmission. main transmission in aircraft.			N/A	N/A	7967.6	20/04/07	N/A	9467.6	N/A	N/A	1223.5	N/A	N/A
Remove freewheel as	sembly from aircraft.		1500	N/A	N/A	7967.6	20/04/07	N/A	9467.6	N/A	N/A	1223.5	N/A	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Las	t Complied V	Vith		Next Due	-		Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Disassemble the free to perform a visual ins and outer shaft. It is n from the outer shaft. stripping of paint is no required.	wheel assembly to the spection of the clutch a ot necessary to remov Remove all sealant and ot	ne extent necessary assembly, inner shaft, e the thrust bearing d clean all parts. The	1500	N/A	N/A	7967.6	20/04/07	N/A	9467.6	N/A	N/A	1223.5	N/A	N/A
Visually inspect clutch assemblies, and beari	ually inspect clutch assembly, inner and outer race shaft emblies, and bearings for serviceability. semble freewheel assembly.				N/A	7967.6	20/04/07	N/A	9467.6	N/A	N/A	1223.5	N/A	N/A
Assemble freewheel a	semble freewheel assembly.				N/A	7967.6	20/04/07	N/A	9467.6	N/A	N/A	1223.5	N/A	N/A
Install freewheel asse	emble freewheel assembly. all freewheel assembly in aircraft.				N/A	7967.6	20/04/07	N/A	9467.6	N/A	N/A	1223.5	N/A	N/A
Thoroughly clean swa solvent (C-304) to rem wipe dry. Visually insp cracks with a 10X map side of swashplate sup assembly. If a crack is inspection.	oroughly clean swashplate support base area with dryclean vent (C-304) to remove all dirt, grease, and oil residue, ther pe dry. Visually inspect full circumference of fillet radius for icks with a 1OX magnifying glass. Pay particular attention to e of swashplate support sembly. If a crack is suspected carry out further dye penetra spection.				N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A
Perform pre-disassem to disassembly. Backla mm) and shall not var measured at three dif reference.	bly backlash check of ash shall be 0.003 to 0. y more than 0.002 incl ferent locations. Recor	the tail gearbox, prior 011 inch (0.08 to 0.28 n (0.05 mm) when rd backlash for later	3000	N/A	N/A	5419.2	N/A	N/A	8419.2	N/A	N/A	175.1	N/A	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Las	t Complied V	Vith		Next Due			Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Remove ctail gearbox cap and oil level sight	input pinion and bear glass.	ings. Remove output	3000	N/A	N/A	5419.2	N/A	N/A	8419.2	N/A	N/A	175.1	N/A	N/A
Inspect tail gearbox sp corrosion and chipped wear patterns.	biral bevel gear and ing I, broken, or worn gea	out pinion gear for r teeth. Inspect gear	3000	N/A	N/A	5419.2	N/A	N/A	8419.2	N/A	N/A	175.1	N/A	N/A
Visually inspect access duplex bearing, input shaft roller alignment scoring, pitting, flakin evidence of overheati	ually inspect accessible areas of tail gearbox input pinion olex bearing, input pinion roller alignment bearing, and ou ift roller alignment bearing and race for roughness, spallin ring, pitting, flaking, broken or damaged retainers, and fo dence of overheating and corrosion. ually inspect tail gearbox studs and dowel pins in case embly for security and damage. Replace damaged				N/A	5419.2	N/A	N/A	8419.2	N/A	N/A	175.1	N/A	N/A
Visually inspect tail ge assembly for security studs and/or dowel pi	earbox studs and dowe and damage. Replace o ins.	l pins in case damaged	3000	N/A	N/A	5419.2	N/A	N/A	8419.2	N/A	N/A	175.1	N/A	N/A
Inspect accessible are corrosion and damage	as of tail gearbox case 2.	and output cap for	3000	N/A	N/A	5419.2	N/A	N/A	8419.2	N/A	N/A	175.1	N/A	N/A
Inspect tail gearbox si condition that may ob indicator for discolora film on painted side. F	ght glass for cracking, oscure level or color of ition, peeling paint, or Remove plastic film if p	crazing, or any oil. Inspect oil level evidence of a plastic present.	3000	N/A	N/A	5419.2	N/A	N/A	8419.2	N/A	N/A	175.1	N/A	N/A
Reassemble tail gearb input pinion assembly gear mesh with spiral	ox. Exercise caution w n if previously removed bevel gear.	hen inserting d, to ensure proper	3000	N/A	N/A	5419.2	N/A	N/A	8419.2	N/A	N/A	175.1	N/A	N/A
Check tail gearbox bac shall be within 0.001 i disassembly backlash	cklash post reassembly nch (0.03 mm) of thos check.	n. Measured values e obtained in pre-	3000	N/A	N/A	5419.2	N/A	N/A	8419.2	N/A	N/A	175.1	N/A	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Las	t Complied V	Vith		Next Due			Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Apply sealant and tou tail gearbox.	ich up finish as require	d post re-assembly of	3000	N/A	N/A	5419.2	N/A	N/A	8419.2	N/A	N/A	175.1	N/A	N/A
50 Hour/12 Month Se main rotor (4 places).	rvicing. Service pitch h	orn trunnion bearing,	50	365	N/A	8244.1	13/02/24	N/A	8294.1	12/02/25	N/A	50.0	359	N/A
50 Hour/12 Month Se bearing, main rotor (2	rvicing. Service pillow Places).	block/trunnion	50	365	N/A	8244.1	13/02/24	N/A	8294.1	12/02/25	N/A	50.0	359	N/A
50 Hour/12 Month Se places).	rvicing. Service grip be	arings, main rotor (2	50	365	N/A	8244.1	13/02/24	N/A	8294.1	12/02/25	N/A	50.0	359	N/A
50 Hour/12 Month Se	rvicing. Service duplex	bearing, swashplate.	50	365	N/A	8244.1	13/02/24	N/A	8294.1	12/02/25	N/A	50.0	359	N/A
50 Hour/12 Month Se places).	rvicing. Service trunnio	on bearing, tail rotor (2	50	365	N/A	8244.1	13/02/24	N/A	8294.1	12/02/25	N/A	50.0	359	N/A
100 Hour/6 Month co requirements.	rrosion control. Carry	out corrosion control	100	182	N/A	8244.1	13/02/24	N/A	8344.1	13/08/24	N/A	100.0	176	N/A
Examine the placards, readable, correctly ap configuration of the h	decals, and markings. plied, and in agreemen elicopter.	Make sure they are nt with the applicable	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Main rotor system. Ve movement, and for co	erify main rotor flap re ondition and security.	straint for freedom of	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Las	Complied V	Vith		Next Due			Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Main rotor system. Ex mast bumping.	amine static stops and	mast for evidence of	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Main rotor system. Ex overall condition. Exa of doublers.	amine main rotor blad mine for cracks, corros	es for cleanliness and ion, and de-bonding	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Main rotor system. Ex cracks, erosion, and d	amine the tip weight c eformation.	ap for corrosion,	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Main rotor system. Ex security.	in rotor system. Examine the mast nut for condition and urity.				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Main rotor system. Ex security.	in rotor system. Examine the mast nut for condition and urity. in rotor system. Examine the trunnion for condition and urity.				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Main rotor system. Ex security.	amine the pillow block	s for condition and	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Main rotor system. Ex	amine the grips for co	ndition and security.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Main rotor system. Ex security.	amine the pitch horns	for condition and	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Main rotor system. Ex security.	amine the static stops	for condition and	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Las	Complied V	Vith		Next Due			Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Main rotor system. Ex evidence of wear dam	amine pitch horn trun lage, and for security.	nion bearings for	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Main rotor system. To 84 to 107 inch-pounds misalignment of anti-s	rque check pillow bloc s (9.4 to 12 Nm). Exami slippage marks (ASB 2(ck retention hardware ine for 06-97-90).	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Main rotor system. Ex inboard of wear sleev corrosion is visible, re corrosion cannot be ro disassemble hub for in	in rotor system. Examine fillet radius of both yoke spindle oard of wear sleeves for evidence of corrosion and damag rosion is visible, remove with fine abrasive pad (C-407). If rosion cannot be removed with pads, or is extensive, assemble hub for inspection and repair.				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Main rotor system. Ex and outboard surface	osion cannot be removed with pads, or is extensive, semble hub for inspection and repair. n rotor system. Examine entire yoke centre section (inboa outboard surfaces) for corrosion and damage.			365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Main rotor system. To metal in accordance v inspection and repair	uch up all repaired are vith applicable main ro instructions.	eas or areas of bare otor hub	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Main rotor system. En shields (up to and incl wear sleeve, strap fitt 308) as required.	sure no sealant voids a uding hub assembly 2(ings, and grip closure.	are visible around 06-011-100-021) or Recoat with sealant (C	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tail rotor assembly. Cl warning stripes.	ean tail rotor blades to	o maintain visibility of	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tail rotor assembly. Ex of attachment.	kamine the yoke for co	ndition and security	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Las	Complied V	Vith		Next Due			Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Tail rotor assembly. Ex security of attachmen	xamine pitch horns for t.	condition and	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tail rotor assembly. Excondition and security	xamine trunnionand tr y of attachment.	unnion cap for	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tail rotor assembly. Ex security of attachmen	otor assembly. Examine counterweight for condition and ity of attachment. otor assembly. Examine tail rotor blades tip block rivet age, corrosion, erosion, and looseness.			365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tail rotor assembly. E damage, corrosion, er	rotor assembly. Examine tail rotor blades tip block rivets nage, corrosion, erosion, and looseness.				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tail rotor assembly. Ex nicks, dents, scratches	l rotor assembly. Examine tail rotor blades tip block rivets nage, corrosion, erosion, and looseness. I rotor assembly. Examine tail rotor blades skins for bulges ks, dents, scratches, or other damage.			365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tail rotor assembly. E	xamine tail rotor blade ea of the chordwise we	s deformed or ights.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tail rotor assembly. Exercise the second s	xamine tail rotor blade or scratches.	s leading edges for	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tail rotor assembly. Ex blade skin mating are particular attention ir	xamine tail rotor blade as for suspected voids a the area of tip blocks	s bonded joints of or cracks. Pay and doublers.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tail rotor assembly. E	xamine tail rotor blade cks in uniball.	s feathering bearings	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Las	t Complied V	Vith		Next Due			Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Tail rotor assembly. Ex security.	kamine the spacer for a	condition and	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tail rotor assembly. Ex security.	xamine the rubber bun	nper for condition and	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tail rotor assembly. Ex security.	xamine the static stop	for condition and	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tail rotor assembly. E	rotor assembly. Examine the nut for condition and securit				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tail rotor assembly. Ex security.	rotor assembly. Examine the nut for condition and securit rotor assembly. Examine the balance wheel for condition urity.				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Forward deck. Remov and security.	e and examine forward	d fairing for condition	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Forward deck. Examin fasteners for conditio	e all fairing and cowlir n.	ng latches and/or	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Forward deck. Examin	e top deck for conditio	on, leaks, and security.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Forward deck. Clean t	ransmission deck.		100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Las	Complied V	Vith		Next Due			Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Forward deck. Examin security, and plenum Examine intake fairing	e air induction cowling for damage, obstruct windows for damage	g for condition and ions, and cleanliness. and cleanliness.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Forward deck. Examin damaged wires and riv	e inlet screen, if instal vets.	led, for missing or	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Forward deck. Examin tubes, ejector tubes, a and security.	e particle separator, if and ejector nozzles for	installed, vortex cleanliness, condition,	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Forward deck. Examin obstruction.	security. ward deck. Examine forward deck drains to ensure absenc truction.				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Forward deck. Examin	ward deck. Examine forward deck drains to ensure absend struction. ward deck. Examine antennas for condition and security.				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Controls. Examine the corrosion, wear, mech	control tubes for signation and second se	s of interference, curity.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Controls. Examine the wear, mechanical dan	links for signs of inter nage, and security.	ference, corrosion,	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Controls. Examine the corrosion, wear, mech	rod and bearings for s nanical damage, and se	igns of interference, curity.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Controls. Examine the corrosion, wear, mech	bellcranks for signs of nanical damage, and se	interference, curity.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Last	Complied V	Vith		Next Due			Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Controls. Examine the corrosion, wear, mech	bellcrank supports for nanical damage, and se	signs of interference, curity.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Controls. Examine the interference, corrosio	attaching bolts and n n, wear, mechanical da	uts for signs of amage, and security.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Controls. Inspect mair for damage, corrosion swaged ends at jam n	n rotor pitch links with a, and cracks. Give part ut or inserts.	a 3X magnifying glass icular attention to	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Controls. Examine N2 condition and security assembly for damage	aged ends at jam nut or inserts. htrols. Examine N2 control linkages on forward deck for idition and security. Pay particular attention to the jacksha embly for damage or evidence of twisting.				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Controls. Verify all con trave throughout rang	ntrols. Examine N2 control linkages on forward deck for dition and security. Pay particular attention to the jacksh embly for damage or evidence of twisting. ntrols. Verify all control tubes and bellcranks for freedom of we throughout range.				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Controls. Examine the security. Examine cop	collective control stic	c for condition and tick, if installed.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Controls. Examine the security Examine copi	cyclic control stick for lot cyclic control stick,	condition and if installed.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Controls. Examine the and security. Examine installed.	anti-torque control po copilot anti-torque co	edals for condition ntrol pedals, if	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Swashplate assembly mast.	and mast. Clean swash	plate assembly and	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Last	Complied V	Vith		Next Due			Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Swashplate assembly from outer ring.	and mast. Disconnect	main rotor pitch links	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Swashplate assembly outer ring	and mast. Disconnect	idler (drive) link from	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Swashplate assembly assembly for axial, rac	and mast. Examine sw lial, and accumulative	ashplate drive wear.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Swashplate assembly mechanical damage a (drive) link from outer	ashplate assembly and mast. Examine idler (drive) link for chanical damage and corrosion damage. Disconnect idler ve) link from outer ring. ashplate assembly and mast. Disconnect boot from swash				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Swashplate assembly outer ring and lift up t deterioration.	ashplate assembly and mast. Examine idler (drive) link for chanical damage and corrosion damage. Disconnect idler ve) link from outer ring. ashplate assembly and mast. Disconnect boot from swash er ring and lift up temporarily. Examine boot for evidence erioration.				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Swashplate assembly bearing, before lubric bearing for condition. evidence of roughness	and mast. Verify condi ation, by rotating oute Bearing must be smoc s, binding, dragging, or	ition of duplex r ring to check duplex oth and show no looseness.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Swashplate assembly sidewalls for wear o	and mast. Examine pir r deterioration.	vot sleeve slot	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Swashplate assembly wear, deterioration, a between pivot sleeve	and mast. Examine piv nd evidence of excessi and swashplate suppo	rot sleeve bearings for ve axial play rt.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Las	t Complied V	Vith		Next Due			Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Swashplate assembly cracks and damage.	and mast. Examine col	lar set bushings for	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Swashplate assembly uniball surfaces for m	and mast. Examine sw echanical damage and	ashplate support and corrosion damage.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Swashplate assembly mechanical damage a lever to sleeve assemb	and mast. Examine col nd corrosion damage. oly pins for condition.	lective lever for Examine collective	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Swashplate assembly Examine for evidence	er to sleeve assembly pins for condition. Ashplate assembly and mast. Examine inner ring for condit mine for evidence of contact with sleeve assembly.				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Swashplate assembly	ashplate assembly and mast. Examine inner ring for condit mine for evidence of contact with sleeve assembly. ashplate assembly and mast. Examine outer ring for condit				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Swashplate assembly diameter of swashpla mast pole.	and mast. Examine the the support for evidence	e upper inside e of contact with the	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Swashplate assembly (drive) link, and main	and mast. Connect and rotor pitch links to out	d secure boot, idler er ring.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Swashplate assembly condition and security	and mast. Examine sw /.	ashplate assembly for	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Swashplate assembly	and mast. Examine ma	ist for condition.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Las	t Complied V	Vith		Next Due			Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Swashplate assembly visible around the swa adhesive (C-307) as re	and mast. Ensure no so ashplate drive collar se quired.	ealant voids are t. Recoat with	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Hydraulic system. Exa leaks, and security.	mine rigid and flexible	fluid lines for chafing,	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Hydraulic system. Exa condition.	mine hydraulic pump f	or leaks and general	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Hydraulic system. Exa general condition	raulic system. Examine hydraulic reservoir for leaks and eral condition				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Hydraulic system. Exa and security.	mine servo actuators f	or leaks, condition,	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Hydraulic system. Exa	mine servo actuator su	pport for condition.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Hydraulic system. If hy visually inspect the tw cracks with a 5X magr support assembly hav web.	ydraulic support 206-0 vo forward webs of the iifying glass. All cracks e originated in the top	01-520-005 installed, support assembly for reported in the edge of the forward	100	365	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A
Hydraulic system. Ver actuators for freedom	ify linkage pivot bolts of rotation and securi	on hydraulic servo ty.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Las	t Complied V	Vith		Next Due			Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Hydraulic system. Exa Button should not be	mine hydraulic filter re extended.	d indicator button.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Hydraulic system. Ver Replenish as required emits bad odor.	ify fluid level in hydrau . Replace fluid if color	llic reservoir. nas changed or if fluid	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Hydraulic system. Exa leaks, condition, and s	mine relief valve and s security.	olenoid valve for	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Transmission and pylo assemblies and bearir	nsmission and pylon assembly. Examine the pylon support				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Transmission and pylo deck fittings and surro	on assembly. Examine to bunding structure	he forward and aft	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Transmission and pylo spherical bearings.	on assembly. Examine t	he deck fitting	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Transmission and pylo bearing for damage a	on assembly. Examine t nd deterioration.	he drag pin and	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Transmission and pylo on deck for condition	on assembly. Examine t and sheared rivets.	he static stop plate	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Transmission and pylo for evidence of main o security.	on assembly. Examine t driveshaft contact and	he isolation mount general condition and	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Las	t Complied V	Vith		Next Due			Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Transmission and pylo hardware for security	on assembly. Examine t of attachment.	the all pylon support	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Transmission and pylo assembly for conditio	on assembly. Examine t n and security.	the transmission	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Transmission and pylo cooler for condition, s	on assembly. Examine t security, and obstruction	the transmission oil ons to air flow.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Transmission and pylo cooler air duct for con chafing damage.	nsmission and pylon assembly. Examine the transmission ler air duct for condition and security. Give special attenti fing damage.				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Transmission and pylo assembly for evidence	on assembly. Examine t e of oil leakage.	the transmission	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Transmission and pylo chafing, damage, and	on assembly. Examine a evidence of leakage.	all fluid lines for	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Transmission and pylo for evidence of contai	on assembly. Examine t mination.	the transmission oil	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Transmission and pylo detectors, including m for accumulated mate	on assembly. Inspect al nast bearing chip detec erial.	l transmission chip tor, if installed,	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Transmission and pylo all transmission chip o	on assembly. Perform o detectors.	operational check of	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Las	t Complied V	Vith		Next Due			Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Transmission and pylo visible around deck m mount. Reapply seala	on assembly. Ensure no ounted supports, fittir nt (C-308) as required.	o sealant voids are ags, and isolation	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Main driveshaft. Exan and couplings for grea	nine the main drivesha Ise leakage.	ft for visual damage	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Main driveshaft. Chec axial movement.	k driveshaft for freedo	m and smoothness of	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Main driveshaft. Exan security.	in driveshaft. Examine main driveshaft attachment bolts f urity. in driveshaft. Inspect TEMP-PLATES for evidence of				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Main driveshaft. Inspo overheating indication excessive discoloratio	ain driveshaft. Examine main driveshaft attachment bolts for curity. ain driveshaft. Inspect TEMP-PLATES for evidence of erheating indication, deterioration, debonding, or cessive discoloration of the epoxy overcoating.				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Electrical. Examine all connectors in the area chafing, general cond	electrical components a of the forward deck a ition, and security.	, wires, cables, and Ind transmission for	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Electrical. Examine all connectors in the pow and security.	electrical components ver plant area for chafi	, wires, cables, and ng, general condition,	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Electrical. Verify illum installed, by depressir on filter head.	ination of airframe fue ng differential switch "	I filter caution light, if press to test" button	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Electrical. Examine all connectors in the area condition, and securit	electrical components a of the aft deck for cha y.	, wires, cables, and afing, general	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Last	Complied V	Vith		Next Due			Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Electrical. Verify navig operation, condition,	ation lights and anti-co and security.	ollision light for	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Electrical. Examine an	tennas for condition a	nd security.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Electrical. Verify opera	ation of cockpit map re	eading light.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Electrical. Verify opera	ation of aft cabin readi	ng lights, if installed.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Electrical. Verify opera	trical. Verify operation of aft cabin reading lights, if insta				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Electrical. Verify opera	ation of defog blowers		100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Electrical. Verify opera audio and mute switc	ation of ENGINE OUT a h, if installed.	nd LOW ROTOR RPM	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Electrical. Verify illum segments by depressin segment, as applicable	ination of all caution p ng test switch, or each e.	anel annunciator light individual light	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Electrical. Verify instru	ument lighting and dim	ming capabilities.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Las	t Complied V	Vith		Next Due			Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Electrical and avionics condition and security	s. Examine battery and y.	vent tubes for	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Electrical and avionics	s. Verify operation of la	anding lights.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Electrical and avionics	s. Verify operation of p	osition lights.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Electrical and avionics in the aft fuselage are	trical and avionics. Examine all optional equipment instal ne aft fuselage area for condition and security.				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Electrical and avionics valve switch.	s. Verify proper operati	ion of fuel sump drain	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Airframe powerplant security.	area. Examine firewall	s for condition and	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Airframe powerplant condition and security exposed upper and lo longerons and titaniu be visible. No indicati	area. Examine engine o y. Using a bright light a wer edges of the joints m engine pan. Edge of ons of corrosion or cra	cowlings and doors for nd mirror, inspect between aluminum sealant coating should cks are acceptable.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Airframe powerplant of loose fasteners and form engine pan.	area. Examine engine l damage. Restore crac	oan area for evidence ked or missing sealant	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Las	t Complied V	Vith		Next Due			Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Airframe powerplant that they are not clog	area. Examine engine ged.	oan drains. Make sure	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Airframe powerplant if installed, for eviden	area. Examine airfram ce of leakage and secu	e fuel filter assembly, rity.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Airframe powerplant condition and security fasteners. On helicopt particular attention fo on engine leg tubes.	area. Examine engine i y. Pay particular attent ters with clamshell aco or correct sealing and f	nount legs for ion for loose ustic mounts, pay or signs of corrosion	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Airframe powerplant condition and security	eners. On helicopters with clamshell acoustic mounts, pa cicular attention for correct sealing and for signs of corros engine leg tubes. rame powerplant area. Examine engine mount fittings for dition and security.				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Airframe powerplant security.	area. Examine rotor br	ake for condition and	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Airframe powerplant connection in engine	area. Examine the prin pan area for condition	nary electrical ground and security.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Powerplant. Examine general condition.	ne powerplant area. Examine engine mount fittings f ion and security. ne powerplant area. Examine rotor brake for conditi y. ne powerplant area. Examine the primary electrical g ction in engine pan area for condition and security. plant. Examine engine inlet bellmouth for obstructio I condition.			365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Powerplant. Examine of attachment.	engine compartment l	nardware for security	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Las	t Complied V	Vith		Next Due			Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Powerplant. Examine	engine for evidence of	fuel or oil leaks.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Powerplant. Examine security, pay particula lines.	all flexible and rigid lir r attention for chafing	es for condition and damage and kinked	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Powerplant. Examine security.	exhaust stacks and cla	mps for condition and	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Powerplant. Examine condition and security	all engine-mounted ac ı.	cessories for	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Powerplant. Examine for condition and secu	verplant. Examine all engine-mounted accessories for Idition and security. verplant. Examine the engine anti-ice actuator and mecha condition and security.				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Powerplant. Inspect for Replace parts as requi	uel control lever bolt h red.	ole and bolt for wear.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Powerplant. Inspect e material.	ngine chip detectors fo	or accumulated	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Powerplant. Perform detectors.	operational check of a	l engine chip	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Powerplant. Examine	engine controls for co	ndition and security.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Las	t Complied V	Vith		Next Due			Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Powerplant. Ensure p	roper throttle friction	of engine N1 control.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Powerplant. Check fre release operation. Rei throttle, if installed.	edom of full throttle g turn to closed position	rip travel and idle . Check copilot	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Powerplant. Ensure fu and maximum stops b dual controls are insta inch (2.00 mm) below	werplant. Ensure fuel control stop lever contact with minir d maximum stops before throttle grip reaches travel limit. al controls are installed, it is permitted for pointer to be 0.0 (h (2.00 mm) below 30° mark when copilot twist grip is used werplant. Rotate pilot throttle grip to idle detent. Fuel con inter must be no more than 0.078 inch (2.00 mm) below th				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Powerplant. Rotate pi pointer must be no m mark on the quadrant	verplant. Rotate pilot throttle grip to idle detent. Fuel con nter must be no more than 0.078 inch (2.00 mm) below th rk on the quadrant. Repeat for copilot throttle grip, if insta				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Powerplant. Verify co motion, and binding.	ntrol linkage for exces	sive looseness, lost	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Powerplant. Verify op governor actuator to full up, Position № go for clearance between collective stick full do DECREASE (extended) and lever stop arm.	nter must be no more than 0.078 inch (2.00 mm) below th k on the quadrant. Repeat for copilot throttle grip, if inst replant. Verify control linkage for excessive looseness, lo ion, and binding. replant. Verify operation of engine N2 control. Position N ernor actuator to full DECREASE (extended), lift collective up, Position N ₂ governor actuator to full INCREASE and ve clearance between governor stop and lever stop arm. Low ective stick full down, position N ₂ governor actuator to fu REASE (extended), verify for clearance between governor lever stop arm.				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Powerplant. Verify N2 motion, and binding.	control linkage for ex	cessive looseness, lost	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Las	t Complied V	Vith		Next Due			Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Drivetrain. Examine fr security.	eewheel assembly for	condition, leaks, and	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Drivetrain. Examine th and security.	ne oil cooler blower im	peller for condition	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Drivetrain. Examine o	il cooler blower impell	er shaft for condition.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Drivetrain. Examine o brackets for grease lea	etrain. Examine oil cooler blower hanger bearings and kets for grease leakage and evidence of overheating.				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Drivetrain. Examine a security.	retrain. Examine oil cooler blower hanger bearings and ckets for grease leakage and evidence of overheating. retrain. Examine aft short shaft (aluminium) for condition urity.				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Drivetrain. Examine a for adequate lubricati	ft short shaft (aluminiu on and freedom of mo	ım) splined adapter vement.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Drivetrain. Torque cho inch-pounds (5.7 to 7.	eck disc pack coupling 9 Nm).	fasteners 50 to 70	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Powerplant. Examine accumulated material	the freewheel unit chi	p detector for	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Powerplant. Examine couplings for conditio	forward short shaft (si n and security.	eel) disc pack	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Las	t Complied V	Vith		Next Due			Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Powerplant. Examine for adequate lubricati	forward short shaft (si on and freedom of mo	teel) splined adapters wement.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Powerplant. Torque c fasteners 50 to 70 incl	heck forward short sha h-pounds (5.7 to 7.9 N	aft disc pack coupling m).	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Airframe aft top deck. attachment fasteners	Remove and examine for condition and secu	aft fairing and ırity.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Airframe aft top deck.	Examine fairing retair	ner for condition.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Airframe aft top deck. condition.	Examine exterior of a	ft deck structure for	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Oil system. Examine o Examine oil cooler pa not blocked.	il cooler area for evide n for cleanliness and m	ence of oil leaks. nake sure the drain is	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Oil system. Examine a and security. Pay part kinked lines.	ll flexible and rigid flui icular attention for ch	d lines for condition afing damage and	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Oil system. Examine o cleanliness.	il cooler core for obstr	uctions and	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Oil system. Examine o condition, and securit	il cooler blower and h y.	ousing for cleanliness,	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Las	t Complied V	Vith		Next Due			Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Oil system. Examine o Check oil for contamin	il tank for leakage, cor nation.	dition, and security.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Oil system. Examine o	il tank supports for co	ndition and security.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Oil system. Examine t condition and security	he seals bonded to the /.	oil cooler for	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Oil system. Examine e installed) for conditio	system. Examine external scavenge oil filter system (STC) (alled) for condition and security.				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tailboom structure. C to remove all grime ar	system. Examine external scavenge oil filter system (STC) called) for condition and security. boom structure. Clean entire tailboom taking particular c remove all grime and dirt.				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tailboom structure. R fairings. Inspect for co	emove tail rotor drives Indition and attachme	haft and gearbox nt of fasteners.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tailboom structure. E deformation, wavines tailboom skin and fair upper quadrant of the the rivets attaching th tailboom.	xamine entire tailboon s, working rivets, and ing. Pay particular atte tailboom. Also pay pa te tail rotor gearbox su	n for cracks, dents, chafing between ention to the left side articular attention to pport to the	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tailboom structure. R	eplace any loose or wo	orking rivets found.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Last	Complied V	Vith		Next Due			Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Tailboom structure. R chafing adhesive tape	eplace removed, dama (C-460) as required.	ged, or worn anti-	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tailboom structure. E condition and security	xamine driveshaft cove y.	er retaining clips for	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tailboom structure. E and security of attach Make sure the vertica inboard rib for cracks.	xamine horizontal stab ment. Verify security c I through bolts can be	ilizer for condition f all attachment bolts. rotated. Inspect the	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tailboom structure. Ir for condition and dan to the vertical fin atta	boom structure. Inspect the vertical fin attachment support condition and damage. Pay particular attention he vertical fin attachment points.				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tailboom structure. E and security.	boom structure. Inspect the vertical fin attachment support condition and damage. Pay particular attention he vertical fin attachment points. boom structure. Examine vertical fin assembly for conditi security.				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tailboom structure. 1 hardware 75 to 95 inc	0. Torque check vertica h-pounds (8.47 to 10.7	al fin attachment /5 Nm).	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tailboom structure. E security, and signs of	xamine vertical fin tail ground contact.	skid for condition,	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tailboom structure. E tailboom and clear an	xamine for open drains y obstructions.	s on lower skin of	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tailboom structure. E	xamine antennas for d	amage and security.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Last	Complied V	Vith		Next Due			Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Tail rotor driveshafts. bearings for excessive overheating.	Examine tail rotor driv grease leakage and ev	reshaft hanger idence of	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tail rotor driveshafts. for condition and secu	tor driveshafts. Examine hanger bearing support brac ndition and security. tor driveshafts. Examine long tail rotor driveshaft or ented driveshafts and aft splined adapter for conditio ty. Check aft splined adapter for adequate lubrication om of movement. tor driveshafts. Torque check disc pack coupling fast 70 inch-pounds (5.7 to 7.9 Nm). Apply torque seal aft ng.			365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tail rotor driveshafts. segmented driveshaft security. Check aft spl freedom of movemen	Examine long tail roto s and aft splined adapt ined adapter for adequ t.	r driveshaft or ter for condition and uate lubrication and	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tail rotor driveshafts. 50 to 70 inch-pounds torquing.	rrity. Check aft splined adapter for adequate lubrication a dom of movement. rotor driveshafts. Torque check disc pack coupling faster to 70 inch-pounds (5.7 to 7.9 Nm). Apply torque seal afte uing. rotor gearbox. Examine tail rotor gearbox support for			365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tail rotor gearbox. Exa condition and security gearbox attachment h	lom of movement. rotor driveshafts. Torque check disc pack coupling faste o 70 inch-pounds (5.7 to 7.9 Nm). Apply torque seal afte uing. rotor gearbox. Examine tail rotor gearbox support for lition and security. Pay particular attention for cracks at box attachment holes.			365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tail rotor gearbox. Exa condition, and securit	amine tail rotor gearbo y.	x for oil leaks,	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tail rotor gearbox. Exa contamination.	amine gearbox oil for e	vidence of	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tail rotor gearbox. Exa	amine tail rotor output	shaft for condition.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tail rotor gearbox. Exa for accumulated mate	amine the tail rotor gea rial.	arbox chip detector	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Last	t Complied V	Vith		Next Due	-		Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Tail rotor gearbox. Pe gearbox chip detector	rform operational cheo	k of the tail rotor	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tail rotor pitch chang control mechanism fo for condition and secu	e controls. Examine the r freedom of travel the urity.	e tail rotor pitch oughout range and	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tail rotor pitch chang pitch change boot for or oil leakage.	e controls. Examine the condition, security, an	e following tail rotor d evidence of grease	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tail rotor pitch chang pitch change lower co bearings	bil leakage. I rotor pitch change controls. Examine the following tail ro ch change lower control mechanism rod for condition of arings I rotor pitch change controls. Examine the following tail ro			365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tail rotor pitch chang pitch change links and security.	e controls. Examine the I spherical bearings for	e following tail rotor condition and	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tail rotor pitch chang pitch change crosshea	e controls. Examine the	e following tail rotor curity.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Tail rotor pitch chang pitch change knurled excessive play.	e controls. Examine the nut for condition and s	e following tail rotor security and liner for	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Instruments. Examine condition, security, ar	instruments and instr nd for correct markings	ument panel for 	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Equipment and furnis	hings. Examine the cat	in floor for condition.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Las	Complied V	Vith		Next Due			Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Equipment and furnis condition and security	hings. Examine the sea /.	t assemblies for	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Equipment and furnis restraints and webbin	hings. Examine passen g fo condition and sec	ger and crew urity.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Equipment and furnis crew inertia reels and reel to confirm the pro mechanism.	aints and webbing fo condition and security. pment and furnishings. Verify operation of passenger a r inertia reels and belt buckles. Pull promptly on each to confirm the proper activation of the reel locking hanism. pment and furnishings. Examine the interior trim for lition and security.			365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Equipment and furnis condition and security	ipment and furnishings. Examine the interior trim for dition and security.			365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Equipment and furnis smoothness of sliding	upment and furnishings. Examine the interior trim for ndition and security. upment and furnishings. Verify sliding windows for safety oothness of sliding.			365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Equipment and furnis condition and security	hings. Examine the ver /.	itilation system for	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Equipment and furnis operation, absence of	hings. Verify nose vent obstructions or debris	s for proper , and open drains.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Equipment and furnis valve switch guard.	hings. Verify for prope	r operation of the fuel	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Equipment and furnis condition, security, ar	hings. Inspect cabin fir nd proper charge.	e extinguisher for	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Las	Complied V	Vith		Next Due			Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Fuselage. Examine for	ward fuselage for conc	lition.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Fuselage. Examine all	cabin door hinges for c	condition and security.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Fuselage. Examine all	cabin door seals for co	ndition and security.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Fuselage. Examine all positive locking and w	elage. Examine all cabin door latches for proper adjustmer sitive locking and wear.				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Fuselage. Examine all security.	cabin door windows fo	or condition and	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Fuselage. Examine all security.	cabin door handles for	condition and	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Fuselage. Drain moist	ure from pitot and stat	ic piping installation.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Fuselage. Examine rer security, if installed. E operation.	novable litter door pos xamine the door post	st for condition and handle for proper	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Fuselage. Examine the condition and security	e external power door a /.	and receptacle for	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Last	Complied V	Vith		Next Due			Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Fuselage. Examine bat attachment for condit	ttery compartment doo tion and security.	or, door seal, and	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Fuselage. Examine concerning condition and security	mponents in battery co y.	ompartment for	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Fuselage. Examine cal installed.	pin entry steps for cond	dition and security, if	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Fuselage. Examine wi	elage. Examine windshields and skylights for condition.				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Fuselage. Examine up	elage. Examine windshields and skylights for condition. elage. Examine upper fuselage for evidence of water leaks				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Fuselage. Examine lov dents, cracks, corrosic condition.	ver fuselage for indicat on, delamination, loose	tion of fuel leaks, e or missing rivets, and	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Fuselage. Examine all Restore sealant as neo	fuselage sealant joints cessary.	for condition.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Fuselage exterior. Exa corrosion, delaminati	mine aft fuselage for on, loose or missing riv	dents, cracks, rets, and condition.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Fuselage exterior. Exa oil leaks.	mine aft fuselage for in	ndications of fuel or	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Las	t Complied V	Vith		Next Due			Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Fuselage exterior. Exa	mine the fuel cap for c	condition and security.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Fuselage exterior. Exa	mine the grounding pl	ug for condition.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Fuselage exterior. Exa freedom from obstruc	mine all fuselage drair tions.	is for condition and	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Fuselage exterior. 6. I inlets and outlets for o obstructions, if install	elage exterior. 6. Examine heater and/or air conditioning its and outlets for cleanliness and absence of tructions, if installed.				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Fuselage exterior. Exa latches for operation,	elage exterior. 6. Examine heater and/or air conditioning ets and outlets for cleanliness and absence of tructions, if installed. elage exterior. Examine baggage compartment door, seal, thes for operation, condition, and security.				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Fuselage exterior. Exa for cracks corrosion, d	mine aft crosstube sup listortion, and loose fa	oport tunnel structure steners.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Fuselage interior. Exa	nine baggage compart	ment for condition.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Fuselage interior. Gair access panels in bagga on the aft right side fu for condition.	n access to inside of af age compartment and Iselage. Examine acces	t fuselage through access panel located s panels and fasteners	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Fuselage interior. Exam leaks.	mine under engine par	n for fuel, oil, or water	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Las	t Complied V	Vith		Next Due			Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Fuselage interior. Exa	mine drain lines for co	ndition and security.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Fuselage interior. Exam corrosion, and condition longeron between aft frame at STA 206.76.	mine aft fuselage long ion. Pay particular atte fuselage frame at STA	erons for cracks, ntion to the upper left 179.92 and fuselage	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Fuselage interior. Exa engine pan structure f fasteners.	mine interface betwee for cracks, corrosion, d	n longerons and istortion, and loose	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Fuselage interior. Insp for condition.	ect engine mount at	tachment structure	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Fuselage interior. Exam on fuselage aft bulkhe bulkhead for cracks. C attachment fittings/ in between the intercos inspection of the uppo	selage interior. Inspect engine mount attachment structu condition. selage interior. Examine the four tailboom attachment fitti fuselage aft bulkhead and tailboom forward lkhead for cracks. Give special attention to the tailboom achment fittings/ intercostals and bolts, and to fasteners tween the intercostals. Particular attention must be given to spection of the upper left fitting.				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Crosstubes. Examine o attaching hardware fo	crosstube retaining str or condition and securi	aps, cushions, and ty.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Crosstubes. Examine f and security.	uselage attachment fi	tings for condition	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Crosstubes. Examine e security, if installed.	electrical bonding strip	s for condition and	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Last	Complied V	Vith		Next Due			Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Crosstubes. Examine f installed.	airings for condition a	nd security, if	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Crosstubes. Examine of condition and security joints around support required.	crosstube riveted or cl 7. Give particular atten 8. Restore sealant (C-2	amped supports for tion to the sealant 51) and paint finish as	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Crosstubes. Examine of particular attention to and/or restore surface	ired. Stubes. Examine crosstubes for corrosion and damage. C cular attention to areas where equipment is mounted. or restore surface protection as required. Stubes. Examine sealant joint at junction with skid tube les. Restore sealant (C-251) and paint finish as required			365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Crosstubes. Examine s saddles. Restore seala	l/or restore surface protection as required. sstubes. Examine sealant joint at junction with skid tube dles. Restore sealant (C-251) and paint finish as required.				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Skid tubes. Examine sl security.	osstubes. Examine sealant joint at junction with skid tube Idles. Restore sealant (C-251) and paint finish as required. d tubes. Examine skid tubes for corrosion, damage, and curity.				N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Skid tubes. Examine sl	kid shoes for condition	and security.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Skid tubes. Examine sl security.	kid tube saddles for co	rrosion, damage, and	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Skid tubes. Examine g condition and security	round handling wheel /.	attachment bolts for	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Ground run. Check oil engine, and tail rotor running helicopter.	level of transmission, gearbox prior to	hydraulic tank,	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Las	t Complied V	Vith		Next Due	-		Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Ground run. Start the NR to check for leaks, parameters are withir helicopter.	helicopter and conduc and ensure all system ı Flight Manual limitati	t ground run at 100% s are operational, and ons. Shut down the	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
100 Hour/12 Month S tail rotor (2 places).	ervicing. Service pitch	change mechanism,	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
200 Hour/ 12 month S transmission oil.	ervicing. Drain and rep	lace main	200	365	N/A	8244.1	13/02/24	N/A	8444.1	12/02/25	N/A	200.0	359	N/A
300 Hour/ 6 Month Se driveshaft206-040-10	Hour/ 6 Month Servicing. Service couplings of main reshaft206-040-100 (2 places).				N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A
Main Driveshaft. Insp wear and corrosion, b corrosion. Inspection lubrication requireme	Hour/ 6 Month Servicing. Service couplings of main reshaft206-040-100 (2 places). In Driveshaft. Inspect inner and outer coupling teeth spline ar and corrosion, boot for condition, shaft for damage and rosion. Inspection is to be accomplished concurrently with rication requirements.				N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A
Lubricate hydraulic pu rotor tachometer spli	ımp driveshaft splines, nes.	oil pump splines, and	300	365	N/A	8244.1	13/02/24	N/A	8544.1	12/02/25	N/A	300.0	359	N/A
Examine N1 and N2 ta and gearbox adapter s shaft splines.	chometer generator sl splines for condition. C	naft splines lean and lubricate	300	365	N/A	8244.1	13/02/24	N/A	8544.1	12/02/25	N/A	300.0	359	N/A
Examine forward shorward shorward shorward and lubricate.	t shaft (steel) splined a	idapters for excessive	300	365	N/A	8244.1	13/02/24	N/A	8544.1	12/02/25	N/A	300.0	359	N/A
Lubricate oil cooler bl bearings and check fo	ower and tail rotor dri r evidence of overheat	veshaft hanger ing.	300	365	N/A	8244.1	13/02/24	N/A	8544.1	12/02/25	N/A	300.0	359	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Las	t Complied V	Vith		Next Due			Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Examine and lubricate driveshaft splined ada	nine and lubricate aft short shaft (aluminum) and aft tail eshaft splined adapters. rotor control tube without nylatron sleeves. To gain acce rol tube, remove tail rotor gearbox fairing and access pa ght side of aft fuselage adjacent to tailboom. Remove ar ect tail rotor control tube for wear at the five areas wher o contacts tailboom fairleads. Remove attachment bolts a e end of control tube through aft end of tailboom and ins wear at five areas where tube contacts fairleads.			365	N/A	8244.1	13/02/24	N/A	8544.1	12/02/25	N/A	300.0	359	N/A
Tail rotor control tube control tube, remove on right side of aft fus inspect tail rotor cont tube contacts tailboon each end of control tu for wear at five areas	veshaft splined adapters. I rotor control tube without nylatron sleeves. To gain access atrol tube, remove tail rotor gearbox fairing and access pan right side of aft fuselage adjacent to tailboom. Remove and pect tail rotor control tube for wear at the five areas where the contacts tailboom fairleads. Remove attachment bolts a th end of control tube through aft end of tailboom and insp wear at five areas where tube contacts fairleads. I rotor control tube without nylatron sleeves. Measure meter of tube just forward and aft of each worn area. Mak asurements in the same plane. Measure tube diameter in to a and compare with previous measurements. If wear does end through tube finish, reinstall tube. If wear in any area ends through dry film lubricant (black color) and the anodio own color), but it is not in excess of 0.004 inch (0.01 mm) of which tube may be rated 190° and reinstalled. If wear				N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A
Tail rotor control tube diameter of tube just measurements in the area and compare wit extend through tube d extends through dry f (brown color), but it is one side, tube may be extends completely an (0.01 mm) on any side wear is found in any c any side, remove the	e without nylatron slee forward and aft of eac same plane. Measure is in previous measureme finish, reinstall tube. If ilm lubricant (black col s not in excess of 0.004 e rotated 180° and rein round tube and does n e, reinstall tube assemb one area in excess of 0. assembly from service.	wes. Measure h worn area. Make all tube diameter in worn ents. If wear does not wear in any area lor) and the anodize l inch (0.01 mm) on stalled. If wear ot exceed 0.004 inch oly end for end. If 004 inch (0.01 mm) on	600	365	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A
Tail rotor control tube tube, parts, and pane	e without nylatron slee Is that were removed.	eves. Install control	600	365	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A
Main driveshaft. Inspo wear and corrosion, b corrosion. Inspection lubrication requirement	ect inner and outer cou loot for condition, and is to be accomplished ents.	upling teeth splines for shaft for damage and concurrently with	600	365	N/A	8244.1	13/02/24	N/A	8844.1	12/02/25	N/A	600.0	359	N/A
Tail rotor control tube gearbox faring and ac	e with nylatron sleeves cess panel. Remove tai	. Remove tail rotor I rotor control tube.	1200	730	N/A	8244.1	02/01/24	N/A	9444.1	01/01/26	N/A	1200.0	682	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Las	t Complied V	Vith		Next Due			Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Tail rotor control tube control tube and nylar contacts fairleads. If n but surface of control reinstalled. If nylatror surface of control tub 0.004 inch (0.01 mm) sleeves. If control tub at any point, or if wea regardless of depth, re	with nylatron sleeves tron sleeves at five are nylatron sleeves are wo tube is unworn, tube is a sleeves are worn thro e, and control tube is on one side only, repla e is worn greater than r extends more than h eplace control tube.	. Inspect tail rotor as where tube orn on one side only, may be rolled 180° and bugh, exposing bare worn not in excess of ice worn nylatron 0.004 inch (0.01 mm) alfway around tube	1200	730	N/A	8244.1	02/01/24	N/A	9444.1	01/01/26	N/A	1200.0	682	N/A
Tail rotor control tube corrosion. Pay particu sleeves.	with nylatron sleeves lar attention to bond l	. Inspect tube for ine at nylatron	1200	730	N/A	8244.1	02/01/24	N/A	9444.1	01/01/26	N/A	1200.0	682	N/A
Tail rotor control tube in tailboom. Reinstall on right side of aft fus	e with nylatron sleeves tail rotor gearbox fairi elage.	. Install control tube ng and access panel	1200	730	N/A	8244.1	02/01/24	N/A	9444.1	01/01/26	N/A	1200.0	682	N/A
Transmission inspection	on.		2250	1825	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A
Inspect all fuel system for proper functioning within the fillet caps a replace the filler caps	n, oil system, and hydra g and sealing. Make su are in good condition. I or replace sealing o-ria	aulic system filler caps re the sealing o-rings Repair or ngs as required.	N/A	365	N/A	8244.1	13/02/24	N/A	N/A	12/02/25	N/A	N/A	359	N/A
Perform a dynamic ba assembly.	lance of the main roto	r hub and blade	N/A	365	N/A	8244.1	13/02/24	N/A	N/A	12/02/25	N/A	N/A	359	N/A
Do an operational che	ck of the fuel low cau	ion system.	N/A	365	N/A	8244.1	13/02/24	N/A	N/A	12/02/25	N/A	N/A	359	N/A
Remove fuel boost pu for debris, water cont	mp assemblies and ins amination, and fungus	pect fuel cell interiors growth.	N/A	365	N/A	8244.1	13/02/24	N/A	N/A	12/02/25	N/A	N/A	359	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Last	Complied V	Vith		Next Due			Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Remove overhead upl access panels. Inspect for cracks and corrosi	nolstery, hat bin, sound engine and transmissi on.	dproofing blanket, and on support structure	N/A	365	N/A	8244.1	13/02/24	N/A	N/A	12/02/25	N/A	N/A	359	N/A
Remove mast nut, fla a bright light, inspect for corrosion and con	o restraint kit, and cap internal surface (intern dition of protective coa	plug. With the use of hal diameter) of mast ating.	N/A	365	N/A	8244.1	13/02/24	N/A	N/A	12/02/25	N/A	N/A	359	N/A
Install cap plug, flap r mast nut 250 to 275 fo hours of flight operati	estraint kit, and mast r oot-pounds (339.0 to 3 on.	ut. 2. Torque check 72 Nm) after 1 to 5	N/A	365	N/A	8244.1	13/02/24	N/A	N/A	12/02/25	N/A	N/A	359	N/A
Main flight control bo bolts/nuts (1 through	in flight operation. in flight control bolts/nuts. Remove main rotor flight cont ts/nuts (1 through 13, Figure 5-2).			730	N/A	8244.1	02/01/24	N/A	N/A	01/01/26	N/A	N/A	682	N/A
Main flight control bo Wipe dry.	lts/nuts. Clean bolts/n	uts with MEK (C-309).	N/A	730	N/A	8244.1	02/01/24	N/A	N/A	01/01/26	N/A	N/A	682	N/A
Main flight control bo corrosion and thread damaged threads, det	lts/nuts. Visually inspe damage. Replace any b ectable wear, and/or o	ect bolts/nuts for olts/nuts that have corrosion.	N/A	730	N/A	8244.1	02/01/24	N/A	N/A	01/01/26	N/A	N/A	682	N/A
Main flight control bo preventive compound installation. Do not ap threads.	lts/nuts. Apply a coati I (C-104) to all bolt sha oply corrosion preventi	ng of corrosion nks prior to ve compound to bolt	N/A	730	N/A	8244.1	02/01/24	N/A	N/A	01/01/26	N/A	N/A	682	N/A
Main flight control bo attaching hardware. T	lts/nuts. Install flight c orque nuts and install	ontrol bolts and new cotter pins.	N/A	730	N/A	8244.1	02/01/24	N/A	N/A	01/01/26	N/A	N/A	682	N/A
Main flight control bo preventive compound exposed threads after	Its/nuts. Apply a coati (C-101) to all bolt hea installation.	ng of corrosion ds, washers, nuts, and	N/A	730	N/A	8244.1	02/01/24	N/A	N/A	01/01/26	N/A	N/A	682	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Las	t Complied V	Vith		Next Due			Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Fuel vent/purge fittin	g filter cleaning.		N/A	730	N/A	8244.1	02/01/24	N/A	N/A	N/A	N/A	N/A	N/A	N/A
Torque Check Main Re	otor Mast Nut.		1	N/A	N/A	8244.1	13/02/24	N/A	8245.1	N/A	N/A	1.0	N/A	N/A
			5			8244.1	13/02/24	N/A	8249.1			5.0		
Torque Check Main Re	otor Pillow Block reten	tion	3	N/A	N/A				3.0	N/A	N/A	N/A	N/A	N/A
Bolts.			8		NA				8.0	NA	N/A	N/A	N/A	17.4
Main Swachplata Tilt	Friction chock		10	N/A	N/A				10.0	N/A	N/A	N/A	N/A	N/A
Main Swashpiate The	Friction check.		25		N/A				25.0		N/A	N/A	N/A	N/A
Dice Dack Coupling To			10	N/A	N/A				10.0	N/A	N/A	N/A	N/A	N/A
Disc Pack Coupling To	rque checks.		25	N/A	N/A				25.0	N/A	N/A	N/A	N/A	N/A
Tail rotor Gearboy Ret	taining Nuts		10	N/A	N/A				10.0	N/A	N/A	N/A	N/A	N/A
			25		N/A				25.0	N/A	N/A	N/A	N/A	174
Engine Mount Hardwa	are Torque Check.		100	N/A	N/A				100.0	N/A	N/A	N/A	N/A	N/A
Tailboom Attachment	Hardware Torque Che	ck.	100	N/A	N/A				100.0	N/A	N/A	N/A	N/A	N/A
Replace Main Transm	ission Oil Filter Elemen	t And Oil.	100	N/A	N/A				100.0	N/A	N/A	N/A	N/A	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	Description			Periodity		Last	t Complied V	Vith		Next Due	_		Remaining	
	Description		Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's	Hrs	Cal	TE's
Main Transmission To	p Case Stud Nut Torqu	e Check.	100	N/A	N/A				100.0	N/A	N/A	N/A	N/A	N/A
Main Transmission To	p Case Paint Touch Up		100	N/A	N/A				100.0	N/A	N/A	N/A	N/A	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
				Periodity		Las	Complied V	Vith		Next Due	-	-	Remaining	
	Description		Hrs	Cal	Сус	Hrs	Cal	Land	Hrs	Cal	Land	Hrs	Cal	Land
				Sc	heduled En	gine Inspecti	ons			- Cu.				
100 Hour Engine Inspect missing bolts, broken or accessory and broken or obvious damage and evi	ion. Inspect the entire er loose connections, secur missing lockwire. Check dence of fuel or oil leaka	ngine for loose or ity of mounting accessible areas for ge.	100	N/A	N/A	8198.2	21/03/23	N/A	8298.2	N/A	N/A	54.1	N/A	N/A
100 Hour Engine Inspect alignment of torque pair apply torque paint.	ion. Inspect all "B" nuts f nt. If missing, loosen "B"	or application and nut, retighten, and	100	N/A	N/A	8198.2	21/03/23	N/A	8298.2	N/A	N/A	54.1	N/A	N/A
100 Hour Engine Inspect sure they are tight, lock screws and rivets. Remo into the compressor inle	ion. Check mounting and wired and in good condit ve all foreign material w .t.	support bolts to be ion. Check security of hich might be drawn	100	N/A	N/A	8198.2	21/03/23	N/A	8298.2	N/A	N/A	54.1	N/A	N/A
100 Hour Engine Inspect lines, and connections for the boost pump on, if av visual condition dictates	ion. Check accessible fue or security, damage or lea railable. Remove, visually	l system components, akage. Accomplish with r inspect and clean if	100	N/A	N/A	8198.2	21/03/23	N/A	8298.2	N/A	N/A	54.1	N/A	N/A
100 Hour Engine Inspect	ion. Inspect Pc filter for p	proper clamping.	100	N/A	N/A	8198.2	21/03/23	N/A	8298.2	N/A	N/A	54.1	N/A	N/A
100 Hour Engine Inspect disassembly or removal bracket, inspect using a detect any signs of crack both of the end fittings a	ion. Inspect Pc filter asse of the Pc filter assembly 10x magnification glass a ss or corrosion, paying pa at their junction with the	mbly, without from themounting Ind a bright light to rticular attention to end walls.	100	N/A	N/A	8198.2	21/03/23	N/A	8298.2	N/A	N/A	54.1	N/A	N/A
100 Hour Engine Inspect Assembly at both ends a particular attention to tl areas beneath the floati contain cracks, corrosion replaced by new parts. V the end with "FILTER" e on the B-nut is adjacent	ion. Remove the Scroll-to and inspect for cracks usi he flared ends of the tub ng ferrules for fretting da n and/or excessive frettin When re- installing the tu toched on the tube or the to the filter.	p-Pc Filter Tube ng 10x power glass. Pay e for cracks, and to the amage. Tubes found to ng damage are to be be assembly, make sure end with 'FILTER END"	100	N/A	N/A	8198.2	21/03/23	N/A	8298.2	N/A	N/A	54.1	N/A	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
			-			-			-			•		
	Description			Periodity		Las	Complied V	Vith		Next Due	•		Remaining	
-	•		Hrs	Cal	Сус	Hrs	Cal	Land	Hrs	Cal	Land	Hrs	Cal	Land
100 Hour Engine Inspect removed and using a 10 compressor scroll for dis permissible.	tion. With the Scroll-to-Po x power glass, inspect th stress/cracks/proper alig	: Tube Assembly still e elbow in the nment. No cracks are	100	N/A	N/A	8198.2	21/03/23	N/A	8298.2	N/A	N/A	54.1	N/A	N/A
100 Hour Engine Inspect governor linkage for free Check security of linkage	tion. Check fuel control a edom of operation, full ti e for loose or worn linkag	nd power turbine ravel and proper rigging. ge and linkage bolts.	100	N/A	N/A	8198.2	21/03/23	N/A	8298.2	N/A	N/A	54.1	N/A	N/A
100 Hour Engine Inspect visible blades and vanes	tion. Inspect compressor ; for foreign object dama	inlet guide vanes and ge.	100	N/A	N/A	8198.2	21/03/23	N/A	8298.2	N/A	N/A	54.1	N/A	N/A
100 Hour Engine Inspect solution as required if o airborn pollutants or wi	tion. Clean compressor w perating in a smoggy area th water alcohol.	ith chemical wash a, conditions with	100	N/A	N/A	8198.2	21/03/23	N/A	8298.2	N/A	N/A	54.1	N/A	N/A
100 Hour Engine Inspect for build-up of contamir pattern. Ultrasonic clear	tion. Visually inspect the nants which could restrict n nozzles if equipment is	water-alcohol nozzles : flow or alter the spray available.	100	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A
100 Hour Engine Inspect water-alcohol injection	tion. Clean the 200 mesh kit).	screen (if equipped with	100	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A
100 Hour Engine Inspect breaks at the antiice a breaks are detected, cho	tion. Inspect the compres ir valve and customer ble eck engine for possible vi	sor scroll for cracks or ed port. If cracks or bration causes.	100	N/A	N/A	8198.2	21/03/23	N/A	8298.2	N/A	N/A	54.1	N/A	N/A
100 Hour Engine Inspect are cocked or backing ou detected, check engine	tion. Inspect for discharge ut of the scroll. If cocked for possible vibration cau	e air tube inserts that or loose inserts are ses.	100	N/A	N/A	8198.2	21/03/23	N/A	8298.2	N/A	N/A	54.1	N/A	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
									1					
	Description		Hrs	Periodity	Cvrc	Last	Complied V	Vith	Hrs	Next Due	Land	Hrc	Remaining	Land
100 Hour Engine Inspect for cracks,damage, dete mirror as necessary. The removed. Perform a Lea discharge tubes and FPI	ion. Visually inspect com rioration or corrosion usi e compressor discharge to k Tec check for an install removed tubes.	pressor discharge tubes ng a bright light and ubes do not have to be ed compressor	100	N/A	N/A	8198.2	21/03/23	N/A	8298.2	N/A	N/A	54.1	N/A	N/A
100 Hour Engine Inspect and proper operation. V unless a problem is dete	ion. Check antiice valve alve need not be remove ccted.	for security, worn parts d or disassembled	100	N/A	N/A	8198.2	21/03/23	N/A	8298.2	N/A	N/A	54.1	N/A	N/A
100 Hour Engine Inspect nuts for looseness, frett required. Check engine f	proper operation. Valve need not be removed or disassembled ess a problem is detected. Hour Engine Inspection. Inspect compressor mount inserts, bolts is for looseness, fretting or oil leakage. Replace or retighten as uired. Check engine for possible vibration causes. Hour Engine Inspection. Clean the female splines of the		100	N/A	N/A	8198.2	21/03/23	N/A	8298.2	N/A	N/A	54.1	N/A	N/A
100 Hour Engine Inspect startergenerator gears generator with mineral Startergenerator Gears	ion. Clean the female spl hafts and the male spling spirits and a soft brush. I shaft Female Spline Inspe	ines of the es of the starter nspect splines. Refer to ection paragraph.	100	N/A	N/A	8198.2	21/03/23	N/A	N/A	N/A	N/A	N/A	N/A	N/A
100 Hour Engine Inspect wear in accordance with inspection is made.	ion. Inspect the starter the Aircraft Manual at t	generator brushes for he same time the spline	100	N/A	N/A	8198.2	21/03/23	N/A	N/A	N/A	N/A	N/A	N/A	N/A
100 Hour Engine Inspect Aeroshell No. 22, or equ generator, make sure to generator driveshaft are	ion. Lubricate acceptable ivalent. Before reinstalla rsional damper members i in hard contact with eac	e splines with grease, tion of the starter s of the starter ch other.	100	N/A	N/A	8198.2	21/03/23	N/A	N/A	N/A	N/A	N/A	N/A	N/A
100 Hour Engine Inspect Note any accumulation Conduct further inspect train/bearings if metal c 39, and 40 below if carb	ion. Remove, inspect, an of metal chips, debris, or ion of the lube system an hips or debris are found. on particles are found.	d clean the oil filter. carbon particles. d/or engine gear See Items 21, 21D, 38,	100	N/A	N/A	8198.2	21/03/23	N/A	8298.2	N/A	N/A	54.1	N/A	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
								-			-	-		_
	Description		Une	Periodity	Gue	Las	t Complied V	Vith	Una	Next Due	Land	11.00	Remaining	Land
100 Hour Engine Inspect	ion. Inspect magnetic ch	p detector plugs.	100	N/A	N/A	8198.2	21/03/23	N/A	8298.2	N/A	N/A	54.1	N/A	N/A
100 Hour Engine Inspect detector plugs and flang	ion. Inspect quick discon red inserts for wear, if ins	nect magnetic chip talled.	100	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A
100 Hour Engine Inspect and inspect the fuel noz	ion. Remove, inspect and zle filter.	l clean the fuel nozzle	100	N/A	N/A	8198.2	21/03/23	N/A	8298.2	N/A	N/A	54.1	N/A	N/A
100 Hour Engine Inspect gasket, without removir (blowouts) can be detec	ion. Check the condition ng bleed valve. Replace g ted.	of the bleed valve asket if air leaks	100	N/A	N/A	8198.2	21/03/23	N/A	8298.2	N/A	N/A	54.1	N/A	N/A
100 Hour Engine Inspect (sheet metal and weld s weld seams in the area fuel nozzle bosses, armp light and mirror as neces Perform a Leak Tec chec OCC's.	ion. Visually inspect the eams) for cracks. Pay par of the igniter plugs, dum it braze patch and adjac ssary. The OCC does not l k for installed OCC's and	outer combustion case ticular attention to the ny plug, drain valves, ent areas. Use a bright nave to be removed. an FPI for removed	100	N/A	N/A	8198.2	21/03/23	N/A	8298.2	N/A	N/A	54.1	N/A	N/A
100 Hour Engine Inspect	ion. Clean the burner dra	in valve.	100	N/A	N/A	8198.2	21/03/23	N/A	8298.2	N/A	N/A	54.1	N/A	N/A
100 Hour Engine Inspect chafing, or cracking of co lockwire.	ion. Inspect the ignition of t	ead for burning, ors and broken	100	N/A	N/A	8198.2	21/03/23	N/A	8298.2	N/A	N/A	54.1	N/A	N/A
100 Hour Engine Inspect gearbox accessory drive removal of bearing cage looseness.	ion. If optional CEB 1165 gear), has been accompl , then inspect bearing ca	, (removal of spare ished without ge retention for	100	N/A	N/A	8198.2	21/03/23	N/A	8298.2	N/A	N/A	54.1	N/A	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	-		-								-			
	Description		Hrs	Periodity	Cvc	Las	Cal	Vith	Hrs	Next Due	Land	Hrs	Remaining	Land
200 Hour Engine Inspect Sundstrand dual elemer 6856250, 6876803.	ion. Perform fuel pump l It pump P/N 6854292, 68	oacklash inspection on 57548, 6877719,	200	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A
300 Hour Engine Inspect a throw away item. It is for signs of contaminant system and clean if nece	ion. Replace the fuel filte not cleanable. Before dis ts. If any are found, inspe essary.	er element. This filter is scarding filter, inspect sct the entire fuel	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
300 Hour Engine Inspect check when a fuel filter NOTE: Applicable to Sun Argo-Tech/TRW manufa	ion. Do a fuel pump bypa is replaced. Idstrand/Pesco and Eaton Ictured pumps only.	ass valve operation	300	N/A	N/A	7967.6	20/04/07	N/A	N/A	N/A	N/A	N/A	N/A	N/A
300 Hour Engine Inspect	ion. Remove, clean and i	nspect engine Pc filter.	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
300 Hour Engine Inspect pressure reducer.	ion. Inspect and clean th	e No. 1 bearing oil	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
300 Hour Engine Inspect internal carbonous depo necessary.	ion. Visually inspect extension. Visually inspect extension of the second state of the	rnal sump. Clean Imp or replace if	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
300 Hour Engine Inspect turbine support. Clean c	tion. Inspect scavenge oil arbonous deposits from	strut in the power strut.	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
300 Hour Engine Inspect nozzle. Clean internal ca	ion. Inspect No. 6 and 7 l irbonous deposits from n	bearing pressure oil ozzle.	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
300 Hour Engine Inspect	ion. Inspect the thermoc	ouple assembly.	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
				Dorioditu		Loc	Complied	Vith	1	Novt Duo	-		Romaining	-
	Description		Hrs	Cal	Cvc	Hrs	Cal	Land	Hrs	Cal	Land	Hrs	Cal	Land
300 Hour Engine Inspect and inspect fuel nozzle f	ion. Remove and disasse ilter assembly. Assemble	mble fuel nozzle. Clean and install fuel nozzle.	300	N/A	N/A	8244.1	13/02/24	N/A	8544.1	N/A	N/A	300.0	N/A	N/A
300 Hour/12 Month Eng blades, and vanes when environment. 10X powe corrosion pit inspection.	ine Inspection. Inspect th operating in an erosive a r magnification is recom	ne compressor case, Ind/or corrosive nended for	300	365	N/A	8244.1	13/02/24	N/A	8544.1	12/02/25	N/A	300.0	359	N/A
500 Hour/12 Month Eng P/N 6846278 and 68713 corrosion.	zine Inspection. Inspect a 38 power turbine outer c	ll uncoated and coated ouplings nuts for	500	365	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A
600 Hour Engine Inspect Sundstrand single-eleme	ion. Check the fuel pump ent pumps for spline wea	driveshaft on the r.	600	N/A	N/A	7967.6	20/04/07	N/A	8567.6	N/A	N/A	323.5	N/A	N/A
600 Hour Engine Inspect functional test per Facet M250 CSL 1164) for aircu filter system. Follow the this recommended inspe	ion. Do the scavenge oil t Service Bulletin No. 090 raft equipped with this ty Facet instructions and ti ection interval each 600 h	filter impending bypass 589 (Ref. Rolls-Royce pe of external scavenge me intervals, or follow iours.	600	N/A	N/A	7967.6	20/04/07	N/A	8567.6	N/A	N/A	323.5	N/A	N/A
600 Hour Engine Inspect Bendix fuel controls P/N prior unless M250-C20 C	ion. Replace the fuel con I 2524552-4 or 2524552-6 EB1089 has been accor	trol filter assembly. (less-5) and nplished.	600	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A
1000 Hour Engine Inspec governor per M250 CEB- from the engine to perfo	ction. Inspect Py port on A-1281. The governor m orm this inspection.	Bendix power turbine ust be removed	1000	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A
1500 Hour Engine Inspec	tion. Replace the fuel co	ntrol filter assembly.	1500	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	-			Poriodity		Lac	Complied	Vith	1	Noxt Duo			Pompining	
	Description		Hrs	Cal	Cvc	Hrs	Cal	Land	Hrs	Cal	Land	Hrs	Cal	Land
1750 Hour Engine Inspec	ction. Carry out HMI.		1750	N/A	N/A	7967.6	20/04/07	N/A	9717.6	N/A	N/A	1473.5	N/A	N/A
100 Hour/6 Month Engi	ne Lubrication. Replace e	ngine oil.	100	182	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A
200 Hour/6 Month Engi	ne Lubrication. Replace e	ngine oil.	200	182	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A
300 Hour/12 Month Eng	ine Lubrication. Replace	engine oil.	300	365	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A
600 Hour/12 Month Eng	ine Lubrication. Replace	engine oil.	600	365	N/A	8244.1	13/02/24	N/A	8844.1	12/02/25	N/A	600.0	359	N/A
100 Hour Engine Inspect check valve.	ion. Inspect and clean th	e turbine pressure oil	600	365	N/A	8244.1	13/02/24	N/A	8844.1	12/02/25	N/A	600.0	359	N/A
100 Hour Engine Inspect Detach the clamp; then at the connector (tee). L elbow only to allow suff of the screen. At assemi 250 lb in. (2328 N.m). 120 lb in. (914 N.m). Ti N.m).	ion. Turbine pressure oil disconnect the power tun oosen the tube coupling icicient movement of the t bly, tighten the connecto Fighten the fireshield elbo ghten the clamp nut to 3	tube screen assembly. rbine pressure oil tube nut at the fireshield sube to enable removal r coupling nut to 200 ow coupling nut to 80 540 lb in. (3.94.5	600	365	N/A	8244.1	13/02/24	N/A	8844.1	12/02/25	N/A	600.0	359	N/A
100 Hour Engine Inspect passage or external surr recommended the exter	tion. Measure the oil flow of the power turbine s rnal sump is not removed	r from the scavenge upport. It is for this check.	600	365	N/A	8244.1	13/02/24	N/A	8844.1	12/02/25	N/A	600.0	359	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
				Periodity		Las	t Complied V	Vith		Next Due	_		Remaining	
	Description		Hrs	Cal	Сус	Hrs	Cal	Land	Hrs	Cal	Land	Hrs	Cal	Land
				Repeti	tive Airwort	hiness Requi	rements						G ui	
	Airworthiness Review.		N/A	365	N/A	8244.1	19/02/24	N/A	N/A	18/02/25	N/A	N/A	365	N/A
Inspec	t main transmission sigh	t glass.	N/A	365	N/A	8244.1	13/02/24	N/A	N/A	12/02/25	N/A	N/A	359	N/A
Inspe	ct tail transmission sight	glass.	N/A	365	N/A	8244.1	13/02/24	N/A	N/A	12/02/25	N/A	N/A	359	N/A
Insp	pect thermal circuit break	ters.	N/A	365	N/A	8244.1	13/02/24	N/A	N/A	12/02/25	N/A	N/A	359	N/A
Inspect first aid kit and have enough calend	Inspect tail transmission sight glass. Inspect thermal circuit breakers. ct first aid kit and contents for condition and ensure all conte ve enough calendar life remaining until next 12 month check		N/A	365	N/A	8244.1	13/02/24	N/A	N/A	12/02/25	N/A	N/A	359	N/A
Inspect fire extinguished if ins	r. Weigh extinguisher to e talled inspect pressure g	ensure no depletion, and auge.	N/A	365	N/A	8244.1	13/02/24	N/A	N/A	12/02/25	N/A	N/A	359	N/A
Carry ou	t cabin Roof Boxbeam in:	spection.	N/A	365	N/A	8244.1	13/02/24	N/A	N/A	12/02/25	N/A	N/A	359	N/A
Carry ou	ut crosstube assembly ins	pection.	300	365	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A
Carry out	fuel distribution system i	nspection.	N/A	730	N/A	8244.1	02/04/24	N/A	N/A	02/04/26	N/A	N/A	773	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date	
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24	
	-		-	-		-		-			-	-		
	Description			Periodity		Las	t Complied V	Vith		Next Due			Remaining	
	Description		Hrs	Cal	Сус	Hrs	Cal	Land	Hrs	Cal	Land	Hrs	Cal	Land
Carry out f	forged stabiliser support	inspection.	100	365	N/A	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
Carry o	ut fuel cell vent decal ins	pection.	300	365	N/A	8244.1	13/02/24	N/A	8544.1	12/02/25	N/A	300.0	359	N/A
Carry ou	It freewheel assembly in	spection.	100	365	PD	8244.1	13/02/24	N/A	8344.1	12/02/25	N/A	100.0	359	N/A
EASA AD 2004	-0009R3 Engine control p	ipe inspection.	110	N/A	N/A	8244.1	13/02/24	N/A	8354.1	N/A	N/A	110.0	N/A	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng Hrs	Eng Starts	Eng TSO	N/A	N/A	N/A	Torque events	Landings	Date			
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	N/A	19/02/24			
				Do	cument Vali	dity										
	Description					Date Of Issue	9			[Date Of Expia	iry				
	Airworthiness Review	Certificate				19/02/2024					18/02/2025	;				
	Certificate Of Airwo	rthiness				17/12/2014				18/02/2025 N/A N/A						
	Certificate Of Regis	tration				21/06/2021			N/A N/A							
	Radio Licence	2				06/01/2022			N/A N/A 31/12/2024							
	Insurance Certifi	cate				08/11/2023					07/11/2024	L				
	Weight And C of G	Report				09/12/2014					N/A					
				Flig	ht Manual S	tatus										
	Description					Date Of Issue					Revision					
	BHT-206B-FM-1 Fligh	t manual				20/12/1972				B	-57, 20/12/2	022				

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng 1 Hrs	Eng 1 Cyc	Eng 1 TSO	Eng 2 Hrs	Eng 2 Cyc	Eng 2 TSO	Torque events	Date
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	19/02/24
			-			-			-	-		
Description	P/N	S/N		Life Limits			Due	-	Insp Type		Remaining	-
		-,	Hours	Calendar	RIN	Hours	Calendar	RIN	., ,, ,	Hours	Days	RIN
Main Rotor Blade	206-010-200-033	A69	5000	N/A	N/A	10567.10	N/A	N/A	Ret	2323.0	N/A	N/A
Main Rotor Blade	206-010-200-033	A74	5000	N/A	N/A	10567.10	N/A	N/A	Ret	2323.0	N/A	N/A
Main Rotor Hub	206-011-100-021	JILM-2156	1200	N/A	N/A	9324.30	N/A	N/A	он	1080.2	N/A	N/A
Trunnion	206-011-113-103	A-8776	4800	N/A	N/A	10872.00	N/A	N/A	Ret	2627.9	N/A	N/A
Main Rotor Grip	206-010-102-121A	HBFS 1206	4800	N/A	N/A	10195.80	N/A	N/A	Ret	1951.7	N/A	N/A
Main Rotor Grip	206-010-102-121A	HBFS 979	4800	N/A	N/A	10195.80	N/A	N/A	Ret	1951.7	N/A	N/A
TT Strap	LORD206-310-004-101	LKFS3473	1200	1460	N/A	9324.30	15/12/25	N/A	Ret	1080.2	665	N/A
TT Strap	LORD206-310-004-101	LKFS3474	1200	1460	N/A	9324.30	15/12/25	N/A	Ret	1080.2	665	N/A
Pin	206-010-123-003	DIF-24944	1200	N/A	N/A	9324.30	N/A	N/A	Ret	1080.2	N/A	N/A
Pin	206-010-123-003	DIFS-24979	1200	N/A	N/A	9324.30	N/A	N/A	Ret	1080.2	N/A	N/A
Fitting Assy	206-011-140-001	DIFS4763	1200	N/A	N/A	9290.00	N/A	N/A	Ret	1045.9	N/A	N/A
Fitting Assy	206-011-140-001	HBFS1900	1200	N/A	N/A	9324.30	N/A	N/A	Ret	1080.2	N/A	N/A
Latch Bolt	206-011-260-103	DI29173	2500	N/A	N/A	10624.30	N/A	N/A	Ret	2380.2	N/A	N/A
Latch Bolt	206-011-260-103	DI29194	2500	N/A	N/A	10624.30	N/A	N/A	Ret	2380.2	N/A	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng 1 Hrs	Eng 1 Cyc	Eng 1 TSO	Eng 2 Hrs	Eng 2 Cyc	Eng 2 TSO	Torque events	Date
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	19/02/24
Swashplate & Support Assy	206-010-450-11	GDJG-10655	4800	N/A	N/A	9707.80	N/A	N/A	ОН	1463.7	N/A	N/A
Collective Lever	206-010-467-001	RE-1853	4800	N/A	N/A	9707.80	N/A	N/A	Ret	1463.7	N/A	N/A
Collective Link	206-010-407-001	TR-138	4800	N/A	N/A	9707.80	N/A	N/A	Ret	1463.7	N/A	N/A
Sleeve Assy	206-010-454-109	A-1103	14400	N/A	N/A	17343.90	N/A	N/A	Ret	9099.8	N/A	N/A
Mast Assy	206.040.002.111	EA IE-57704	3000	N/A	N/A	9456.10	N/A	N/A	ОН	1212.0	N/A	N/A
	200-040-002-111	1731-37704	1500	N/A	N/A	9467.60	N/A	N/A	Insp	1223.5	N/A	N/A
Main Rotor Transmission	206-040-002-025	RKW 10242	4500	N/A	N/A	11667.90	N/A	N/A	ОН	3423.8	N/A	N/A
		BKW 10343	1500	N/A	N/A	9467.60	N/A	N/A	lnsp	1223.5	N/A	N/A
Freewbeel Assy	206 040 270 2	BMB 10470	3000	N/A	N/A	8869.70	N/A	N/A	ОН	625.6	N/A	N/A
Freewieer Assy	200-040-270-3	BIND-10470	1500	N/A	N/A	9467.60	N/A	N/A	Insp	1223.5	N/A	N/A
Clutch	CL42250-1	FD 9500	3000	N/A	N/A	8564.20	N/A	N/A	Ret	320.1	N/A	N/A
Servo Actuator Support	206-001-520-005	13333	o/c	o/c	o/c	N/A	N/A	N/A	N/A	N/A	N/A	N/A
Hydraulic Servo	206-076-031-013	5526	3600	N/A	N/A	11567.60	N/A	N/A	ОН	3323.5	N/A	N/A
Hydraulic Servo	206-076-031-13	1745	3600	N/A	N/A	9562.80	N/A	N/A	ОН	1318.7	N/A	N/A
Hydraulic Servo	206-076-031-013	3685	3600	N/A	N/A	11573.80	N/A	N/A	ОН	3329.7	N/A	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng 1 Hrs	Eng 1 Cyc	Eng 1 TSO	Eng 2 Hrs	Eng 2 Cyc	Eng 2 TSO	Torque events	Date
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	19/02/24
Hydraulic Pump & Reservoir	206-076-022-5	CC21-2424	3600	N/A	N/A	9238.10	N/A	N/A	ОН	994.0	N/A	N/A
Tail Rotor Blade	206-016-201-137	CS 3077	2500	N/A	N/A	9742.60	N/A	N/A	Ret	1498.5	N/A	N/A
Tail Rotor Blade	206-016-201-137	CS 3101	2500	N/A	N/A	9742.60	N/A	N/A	Ret	1498.5	N/A	N/A
Tail Rotor Hub Assy	206-011-810-125	AAG 00006	2500	N/A	N/A	8362.20	N/A	N/A	ОН	118.1	N/A	N/A
Tail Rotor Yoke	206-011-819-101	AFS 4040	5000	N/A	N/A	10862.20	N/A	N/A	ОН	2618.1	N/A	N/A
Tail Rotor Tranmission	206-040-400-11	ALO-51150	6000	N/A	N/A	11419.20	N/A	N/A	ОН	3175.1	N/A	N/A
Lower Collective Tube	206-001-194-001	TY 3315	4800	N/A	N/A	10779.40	N/A	N/A	Ret	2535.3	N/A	N/A
Starter Generator	23032-010	958	1000	N/A	N/A	8424.30	N/A	N/A	ОН	180.2	N/A	N/A

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng 1 Hrs	Eng 1 Cyc	Eng 1 TSO	Eng 2 Hrs	Eng 2 Cyc	Eng 2 TSO	Torque events	Date
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	19/02/24
Description	P/N	s/N		Life Limits			Due		Action		Remaining	
	.,	5/14	Hours	Calendar	Cycles	Hours	Calendar	Cycles	Required	Hours	Days	Cycles
			1	Compress	or Section	1	1	I	1		1	
Compressor Assembly	6890550	CAC-33021	3500	N/A	N/A	11468.0	N/A	N/A	он	3223.9	N/A	N/A
1st Stage Wheel	23057111	E92403	o/c	O/C	o/c	N/A	N/A	N/A	N/A	N/A	N/A	N/A
2nd/3rd Stage Wheel	23057112	KR77710	o/c	o/c	o/c	N/A	N/A	N/A	N/A	N/A	N/A	N/A
4th Stage Wheel	23057114	E81116	o/c	o/c	o/c	N/A	N/A	N/A	N/A	N/A	N/A	N/A
5th Stage Wheel	23057115	KR51415	o/c	O/C	O/C	N/A	N/A	N/A	N/A	N/A	N/A	N/A
6th Stage Wheel	23060416	KR36329	o/c	o/c	o/c	N/A	N/A	N/A	N/A	N/A	N/A	N/A
Impeller	23079638	KR121641	3550	o/c	9150	11518.0	N/A	9225	Ret	3273.9	N/A	8593
Coupling Adappter	23079637-1	GK556493	o/c	O/C	O/C	N/A	N/A	N/A	N/A	N/A	N/A	N/A
				Turbine	Section							
Turbing Assembly	22028241	CAT-28008D	3500	N/A	N/A	9717.6	N/A	N/A	ОН	1473.5	N/A	N/A
Turbine Assembly	25050241	CA1-38098P	1750	N/A	N/A	9718.0	n/A	N/A	нмі	1473.9	N/A	N/A
1St Stage Wheel	M250-10223	X614772	1775	N/A	3000	9743.0	N/A	3075	Ret	1498.9	N/A	2443
2nd Stage Wheel	23073854	X642096	1775	N/A	3000	9743.0	N/A	3075	Ret	1498.9	N/A	2443
3rd Stage Wheel	23001967	HX73265	4550	N/A	6000	11036.3	N/A	3853	Ret	2792.2	N/A	3221

A/C Reg	A/C S/N	Eng S/N	A/C TT	AC Datcon	Eng 1 Hrs	Eng 1 Cyc	Eng 1 TSO	Eng 2 Hrs	Eng 2 Cyc	Eng 2 TSO	Torque events	Date
G-IBIG	2202	CAE-830068	8244.1	N/A	4135.9	632	N/A	N/A	N/A	N/A	N/A	19/02/24
Description	P/N	s/N	Life Limits				Due		Action	Remaining		
	.,	0,	Hours	Calendar	Cycles	Hours	Calendar	Cycles	Required	Hours	Days	Cycles
4th Stage Wheel	6853279	HX68695	4550	N/A	6000	10513.7	N/A	1480	Ret	2269.6	N/A	848
Turbine Tie Bolt	23068265	NC88250	N/A	N/A	9000	N/A	N/A	9075	Ret	N/A	N/A	8443
2nd Stage Nozzle Diaphragm	23084419	MA227185	1775	N/A	3000	9743.0	N/A	3075	Ret	1498.9	N/A	2443
				Gearbox	Section							
Gearbox Assembly	6877180	CAG-30038	o/c	o/c	O/C	N/A	N/A	N/A	N/A	N/A	N/A	N/A
				Ancil	aries							
Fuel Control Unit	M250-10816	321765	2500	N/A	N/A	9426.6	N/A	N/A	ОН	1182.5	N/A	N/A
Power Turbine Governor	M250-10847	24248	2000	N/A	N/A	9214.4	N/A	N/A	он	970.3	N/A	N/A
Fump Pump (Argo- Tech)	6899253	T108052	4000	N/A	N/A	10047.9	N/A	N/A	ОН	1803.8	N/A	N/A
Fuel Nozzle	23077068	AG29301	2500	N/A	N/A	10468.0	N/A	N/A	ОН	2223.9	N/A	N/A
Bleed Valve	23053176	FF33094	1500	N/A	N/A	9306.4	N/A	N/A	ОН	1062.3	N/A	N/A