

Service Bulletin Compliance Record

Company:
Category: Airframe
Manufacturer: Cirrus Design Corp.
Model: SR20

Position:
P/N:
S/N: 1964

Aircraft Registration No: G-FREY
ATP Revision: 09/02/2021

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2X-52-07 R4 24/01/2008	ATA 52-10: Passenger and crew doors cabin door rod end replacement	-- Hrs: -- C: --	N/A by S/N	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
2X-78-07 R1 18/12/2007	[Recurring] ATA 78-20: Noise suppressor SR20 heat exchanger inspection	-- Hrs: -- C: --	N/A by S/N	Yes	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
2X-27-14 R3 10/10/2007	ATA 27-00-Flight controls rudder-Aileron interconnect modification	-- Hrs: -- C: --	N/A by S/N	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
2X-32-13 R1 16/05/2006	ATA 32-42: Brakes-wheel conversion and brake upgrade	-- Hrs: -- C: --	N/A by S/N	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
2X-32-14 R1 17/02/2006	ATA 32-10: Main landing gear- main landing gear fairing modification for inspection hole	-- Hrs: -- C: --	N/A by S/N	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
2X-25-17 R1 20/01/2006	ATA 25-10:Cabin-crew seat break-over pin replacement	-- Hrs: -- C: --	N/A by S/N	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA

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2X-28-04 R1 14/11/2005	ATA 28-20: Distribution fuel line chafing	-- Hrs: -- C: --	N/A by S/N	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
2X-05-06 R4 05/05/2005	ATA 25-10: Cabin-crew seat recline lock installation	-- Hrs: -- C: --	N/A by S/N	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
2X-25-06 R4 05/05/2005	ATA 25-10:Cabin-crew seat recline lock installation	-- Hrs: -- C: --	N/A by S/N	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
A2X-25-08 22/06/2004	ATA 25-10: seats-crew seat break-over bolt adjustment	-- Hrs: -- C: --	N/A by S/N	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
A20-27-06 20/09/2002	ATA 27-00-Trim cartridge self-locking nut replacement	-- Hrs: -- C: --	N/A by S/N	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
20-95-05 10/07/2002	Replacement of CAPS activation cable	-- Hrs: -- C: --	N/A by S/N	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
20-95-04 10/07/2002	Replacement of CAPS activation handle bracket	-- Hrs: -- C: --	N/A by S/N	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
20-95-03 10/06/2002	Replacement of CAPS handle access cover	-- Hrs: -- C: --	N/A by S/N	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
20-01-00 25/02/2002	ATA 01-00: Airplane General- Gross weight increase	-- Hrs: -- C: --	N/A by S/N	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA

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20-55-06 27/11/2001	ATA 55-00: Stabilizers-SR20 stabilizer rivet inspection/ replacement	-- Hrs: -- C: --	N/A by S/N	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA

Category: Engine

Position:

ATP Revision: 09/02/2021

Manufacturer: Continental Motors

P/N:

Model: IO-360-ES

S/N: 360662

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
98-9C 17/07/2013	Time Between Overhaul Periods	-- Hrs: -- C: --	Overhaul Information	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
MSB00-6D 19/11/2010	Slick Service Bulletin SB1-00C	-- Hrs: -- C: --	N/A - TCM Magnetos Fitted	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
MSB09-8A 04/12/2009	Inspection and Removal of Certain TCM Engine Hydraulic Lifters	-- Hrs: -- C: --	N/A by DOM	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
MSB09-8 03/11/2009	Inspection and removal of certain TCM engine hydraulic lifters	-- Hrs: -- C: --	N/A by DOM	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
SB03-3 28/03/2003	[Recurring] Differential pressure test and borescope inspection procedures for cylinders	-- Hrs: -- C: --	Carried out 10/02/2021 @ 1642.5 hours REPETITIVE ON CAFAM RGV Aviation Ltd	Yes	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
98-9A 28/03/2003	Time between overhaul periods	-- Hrs: -- C: --	Overhaul Information	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA

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MSB00-5C 10/10/2000	CRANKSHAFT MATERIAL INSPECTION	-- Hrs: -- C: --	N/A by DOM	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
MSB00-6A 08/06/2000	*Slick service bulletin SB 1-00A	-- Hrs: -- C: --	N/A - TCM Magnetos Fitted	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
MSB00-5B 25/05/2000	Superseded by MSB00-5C	-- Hrs: -- C: --	xxxxxx	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
MSB00-5A 28/04/2000	Superseded by MSB00-5B	-- Hrs: -- C: --	xxxxxx	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
MSB96-10 15/08/1996	[Recurring] CRANKSHAFT ULTRASONIC INSPECTION	-- Hrs: -- C: --	At Crankshaft Removal	Yes	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
CSB96-8 25/06/1996	REPLACEMENT OF AIRMELT (NON- VAR) CRANKSHAFTS	-- Hrs: -- C: --	Overhaul Information	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
SIL648 18/10/1994	OPTIONAL CONVERSION TO "SHOWER-OF-SPARKS" IGNITION	-- Hrs: -- C: --	N/A by DOM	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
SB94-8 14/09/1994	[Recurring] Superseded by MSB94-8A	-- Hrs: -- C: --	xxxxxx	Yes	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
CSB94-1 01/02/1994	TELEDYNE CONTINENTAL MOTORS IGNITION SYSTEMS SERVICE BULLETIN-NO. CSB641	-- Hrs: -- C: --	N/A by DOM	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA

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639 01/03/1993	Proper installation of impulse couplings	-- Hrs: -- C: --	N/A by DOM	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
M92-4 R1 05/02/1992	IO AND L/TSIO 360 ROCKER SHAFT STUD INSPECTION	-- Hrs: -- C: --	N/A by DOM	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
M92-4 10/01/1992	Io And L/Tsio 360 Rocker Shaft Stud Inspection	-- Hrs: -- C: --	N/A by DOM	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
M88-10 24/08/1988	CONTAMINATED FUELS	-- Hrs: -- C: --	N/A by DOM	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
M88-4 15/01/1988	TCM PRO-TECH OIL FILTER	-- Hrs: -- C: --	N/A by DOM	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
M84-15 21/12/1984	Superseded by SB03-3	-- Hrs: -- C: --	xxxxx	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA

Category: Propeller

Position:

ATP Revision: 09/02/2021

Manufacturer: Hartzell Propeller

P/N:

Model: PHC-J3YF-1

S/N: FP6640B

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040A 22/12/2010	TURBOCHARGER CHRA CONTAMINATION	-- Hrs: -- C: --	N/A by TIS	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA

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HC-SB-61-136 RI 25/04/2003	Propellers-blades-additional overhaul and inspection procedures for single- and double-shoulder blade bores	-- Hrs: -- C: --	Propeller Overhauled in 2017	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
HC-SB-61-118 RF 15/08/2002	Blades-rework procedures for "Y" shank aluminum blade bearing retention radius	-- Hrs: -- C: --	Propeller Overhauled in 2017	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
165E 21/01/1994	[Recurring] INSPECTION FOR CRACKS IN CERTAIN THREE BLADE "Y" SHANK ALUMINUM HUBS	-- Hrs: -- C: --	Propeller Overhauled in 2017	Yes	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
61S 10/12/1993	[Recurring] Overhaul Periods And Service Life Limits For Hartzell Propellers And Governors	-- Hrs: -- C: --	Overhaul Information	Yes	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
136H 12/03/1993	Superseded by HC-SB-61-136 RI	-- Hrs: -- C: --	xxxxxx	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
136G 15/11/1991	ADDITIONAL OVERHAUL AND INSPECTION PROCEDURES FOR SINGLE AND DOUBLE-SHOULDER BLADE PILOT TUBE BORES	-- Hrs: -- C: --	Propeller Overhauled in 2017	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
118D 25/03/1991	Blades-rework procedures for "Y" shank aluminum blade bearing retention radius	-- Hrs: -- C: --	Propeller Overhauled in 2017	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
136F 10/08/1990	ADDITIONAL OVERHAUL AND INSPECTION PROCEDURES FOR SINGLE AND DOUBLE-SHOULDER BLADE PILOT TUBE BORES	-- Hrs: -- C: --	Propeller Overhauled in 2017	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
136E 14/10/1988	Superseded by 136F	-- Hrs: -- C: --	xxxxxx	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA

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159B 23/05/1986	Propellers-Propellers- Inspection procedures on all C-3-() double shoulder clamps	-- Hrs: -- C: --	Propeller Overhauled in 2017	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
136A 12/07/1983	Superseded by 136B	-- Hrs: -- C: --	xxxxxx	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
118C 13/05/1983	Superseded by 118D	-- Hrs: -- C: --	xxxxxx	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
136 26/08/1982	Superseded by 136A	-- Hrs: -- C: --	xxxxxx	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
136B 01/01/1980	Superseded by 136C	-- Hrs: -- C: --	xxxxxx	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
118B 28/11/1977	Superseded by 118C	-- Hrs: -- C: --	xxxxxx	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA
118A 15/02/1977	MANDATORY OVERHAUL PROCEDURES FOR ALL HARTZELL MANUFACTURED "Y" SHANK PROPERLLER BLADES	-- Hrs: -- C: --	Propeller Overhauled in 2017	No	D: -- Hrs: -- C: --	UK.145.00215 - R82Y530X/EASA - FAA